



This document gives a detailed summary of the new features and improvements of FEM-Design version 17.

We hope you will enjoy using the program and its new tools and possibilities. We wish you success.

StruSoft, Developer Team

# Legend



Pay attention / Note



Useful hint



Example



Clicking left mouse button



Clicking right mouse button



Clicking middle mouse button

#### What is new in FEM-Design 17?

The list of all new major features and improvements is following:

- Correct model tool
- Plastic calculation of trusses, supports and connections
- Post-tensioned cable
- Fire design for steel bars
- Pile
- Composite sections for beams, columns and piles
- Wall buckling
- Properties dialogs available from the quick menu
- Joint stiffness calculation
- Displaying relevant load combination for max. of combinations results
- Fully rigid diaphragm
- Colour palette for displacement-like results in 3D modules
- New tables to list and new listing options
- Wall corbel
- Joint library and many more improvements of steel joints
- Notional load
- Manual position numbering
- Improved documentation module

#### What is new in FEM-Design 17.1?

- Construction stages (Phase 1)
- Improved storey management
- Run Script
- Option to hide extra information in the numeric results

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# 1. Tools

# 1.1. Correct model

*Correct model* tool is a revolutionary function to fix the models having geometrical errors, in semiautomatic way, with supervision of the user. This tool helps to fix the following types of errors in the model:

- object multiplication and overlap
- geometrically incorrect objects (very small region parts, very small angles, divided region edges, etc.)
- incorrectly positioned objects

*Correct model* tool is the replacement of *Object merge* (in earlier versions), which is no more available in version 17. Correct model tool is a **semi-automatic system** which necessarily **needs user-interaction**. It draws attention to the possible errors of the model and in most cases also offers solution(s), but the User has to decide whether it is really an error or not, and if it is, how to fix it.

*Correct model* can be launched from *Tools/Correct model* or from the toolbar (



After launching Correct model, User needs to select the elements to be checked.

Sel.		Correction		Tolerance [m]	Fixed	^
	Delete identical copie	es			0	1
	Fix overlaps				0	
	Chamfer sharp angle	s		0.100	0	1
	Fix small areas and li	nes		0.100	0	
	Merge region lines			0.010	0	
	Align to structure gri	d		0.100	0	1
	Stretch to structure	grid		0.100	0	
	Align regions			0.100	0	1
	Stretch to crossing r	egions		0.100	0	L
	Stretch regions in pla	ane		0.100	0	L
	Stretch lines			0.100	0	L
	Align points			0.100	0	Г
						V
Select a	all Clear all				Marker	
Start	Ignore	Mark	Fix	Show again		
Break	Ignore all	Mark all	Fix all		Auto	

Fix object multiplication and overlap

Fix geometrically incorrect objects

Fit objects to storeys, axes and reference planes

Fit objects to each other

Then in the pop-up dialog, task(s) should be selected. Using the buttons *Select all* and *Clear all*, all correction tasks can be selected or cleared in one click.



After pressing *Start,* the correcting process begins. It runs row by row and the current step turns green. The view is focused on the incorrect object (part).



The table below contains the meaning of the flashing red object depending on the current correction task:

Correction task (type)	What is flashing?
Delete identical copies	object to be deleted
Fix overlap	current object and its modified version are both flashing
Chamfer sharp angles	the part of the region to be removed
Fix small areas and lines	the part of the line/region to be removed
Merge region lines	the region line to be merged
Fixing geometrically incorrect objects	the object to be fixed
Fit objects to storeys, axes and reference planes	the suggested new position of the
Fit objects to each other	incorrectly placed object

In order to decide what to do with each object, the User has the following choices:

- Ignore: the object will not be modified
- Mark: the object will be marked according to the colour and text set in the Marker dialog
- *Fix:* the object will be deleted/modified according to the program's suggestion. The number in the table in *Fixed* cell of the current correction task's row is increased by one. The fixed object turns into green.

*Ignore all, Mark all, Fix all* acts on all further incorrect objects found by the **current correction task** (highlighted by green in the table).

After one of the abovementioned buttons is pressed, the next incorrect object – if exists – gets into focus and is highlighted

Show again focuses on the current objects. It can be useful if the User gets lost in a large model.

Break cancels the correction process but the previously performed modifications are kept.

Once the correction process is finished (or cancelled) the modifications can be applied to the model by pressing either *Apply* or *OK*. In case of OK the dialog closes. All modifications can be discarded by pressing *Reset* if they were not yet applied on the model.

X	Align points			0.100	0
Select all	Clear all				- Marker
Start Break	Ignore Ignore all	Mark Mark all	Fix Fix all	Show again	Auto
Settings >	>	ОК	Cancel	Apply	Reset
Settings >	> Save as defau Load default		Cancel	Apply	Reset

Settings commands let the user save/load the selected correction tasks and their tolerances. A small but quite complex example at the end of this chapter shows in detail how to use *Correct model* tool.

Using *Auto* button and applying its modification on the model is not recommended, because in practice in most cases there is no exact solution:

- fixing an error may generate or eliminate other problems
- fixing the same error in different ways may generate/eliminate different problems
- consequently, the correcting process can be iterative

However, Auto is useful to get an estimation of errors and their type in the model.

See detailed description of each correction task as follows.

# Delete identical copies

If more identical element exists in the same position, it deletes all identical copies, and counts as one correction in the *Fixed* column on the *Correct model* dialog.

This is not working for intermediate sections, post-tensioned cables and corbels.

This is not working for isolated and wall foundations' regions and corbels.



It fixes overlapping regions and lines. Existing objects can be shrunk or erased, but function will not generate new objects (e.g. a region can not fall apart).

Overlapping is allowed for the Loads.

Sharp angles

Sharp corners under 10° will be fixed according to the given tolerance. The highlighted part of object will be removed. If an element has smaller dimension than the given tolerance, the whole element will be removed.

### Fix small areas and lines

Objects within the tolerance range will be removed. It will fix small holes, long narrow areas etc.

### Merge region lines

Merge region lines function merges two lines/curves of a region into one line/curve if they are in the range of tolerance. Too flat curves may be replaced by line. The figure below shows examples how the tolerance is measured.

Align to structural grid













This function aligns objects to structural grids (*axes, storeys* or *reference planes*). Alignment is done by orthogonal projection of object's checkpoints to the structural grid within the tolerance. The checkpoints are shown below for line, arc and circle:





Once an object is placed on a grid by *Correct model* tool, it will not be detached from it by any next correction step.



*Correct model* tool does not check structural grid for possible errors, like axes too close to each other or not perfectly parallel.

### Stretch to structural grid

This function will stretch regions and lines to the intersection of axes within given tolerance.





If the geometry of the object is incorrect (e.g. shell is not laying in plane) this function cannot be performed.

#### Align regions

It aligns a region (projection) to other region's plane that is within the given tolerance both in direction perpendicular to and parallel with the region's plane.





#### Stretch to crossing regions

This function will stretch objects to the crossing regions within given tolerance.



#### Stretch regions in plane

Regions laying in same plane will be stretched to each other within given tolerance.

#### Stretch regions in plane

Lines laying within given tolerance to other objects will be stretched.

#### <u>Align points</u>

Points (objects) will be aligned to closest object within tolerance.





- Correct model cannot handle:
  - peak smoothing regions
  - connections (point, line, surface)
  - bar, shell components
  - building cover
  - it doesn't work in Analysis and Design tabs
- Only the visible objects can be modified.
- Columns and walls need to be vertical and will stay vertical after the modifications as well.
- Isolated, wall foundation and foundation slab need to be and remain horizontal.
- Pile can be placed in any direction, but cannot be horizontal.
- Corbel is not handled by *Correct model* and cannot be used as an object to fit other objects to.



Correct model works in multiple-window mode as well.









The following example shows the process of fixing geometrical errors of a small model with different types of errors:



The example file can be downloaded from here.

Let's launch *Correct model* tool. Select all structural elements by Ctrl+A (or box), then all correction tasks in the dialog by clicking on **Select all**. To get a quick overview of what kind of problems occur in the model, just click on **Auto** button.

rect mo	odel	11	[
Sel.	Correction	Tolerance [m]	Fixed A
х	Delete identical copies		0
x	Fix overlaps		0
x	Chamfer sharp angles	0.100	0
х	Fix small areas and lines	0.100	0
х	Merge region lines	0.010	0
х	Align to structure grid	0.100	0 E
х	Stretch to structure grid	0,100	0
х	Align regions	0.100	0
x	Stretch to crossing regions	0.100	0
х	Stretch regions in plane	0.100	0
х	Stretch lines	0.100	0
X	Align points	0.100	0
Select all	Cear al	-	- Marker
Start	Johore Mark Fix	Show again	
Break	Iprore al Mark al Pix al		Auto
ettings >	OK Cancel	Apply	Reset

The process searches for all errors and fixes them in a way that most probably is not always the way that the user intends to fix them However, this way User can get a good estimation on the type and amount of errors (by looking at the values in the *Fixed* column).

So, in this case, all together about 15 errors should be fixed.

Now click on **Reset** to discard all modifications done by the automatic process.



Now click on **Start** to see and decide about all the errors one by one.



The doubled support in the bottom left corner is found first. Now click on **Fix** to eliminate it.



The number in the *Fixed* column of "Delete identical copies" task is set to 1.

Overlap of the large plate and the small plate over the column is found as next error. The large plate is blinking, so by clicking on **Fix**, that one would be modified. However, we know that it is the small region that is incorrect, so this time let's click on **Ignore** instead.





Now the small plate is blinking, so by clicking on **Fix**, it is modified and fit into the large plate.

The corrected small plate is displayed in green contour. The number in the *Fixed* column of "Fix overlaps" task is set to 1.



The next problem is found by "Fix small areas and lines" task: a very small region part. Let's remove it by clicking on **Fix**.



The removed small region part is displayed by dashed line, and number of fixed objects for "Fix small areas and lines" task is set to 1.

The next problem, a very small hole is highlighted. We decide not to delete it but deal with it later, so let's mark it. First let's click on the **Marker** and set the marking text, colour and font type. After clicking OK in the Marker dialog, let's click on the **Mark**.

~ [	Correct mod	lel			8
	Sel.	Correction	Tolerance [m]	Fixed ^	
	X	Delete identical copies		1	
	X	Fix overlaps		1	
	X	Chamfer sharp angles	0.100	0	Marker
	Х	Fix small areas and lines	0.100	1	
	X	Merge region lines	0.010	U	Comment Too small hole
	X	Align to structure grid	0.100	0	Comment Too small hole
	X	Stretch to structure grid	0.100	0	Colour
	X	Align regions	0.100	0	
	X	Stretch to crossing regions	0.100	0	Font AaBbYyZz →
	X	Stretch regions in plane	0.100	0	······································
	X	Stretch lines	0.100	0	3.
	X	Align points	0.100	0	OK Cancel
	Select a	Clear all 4.	1.	Marker	
	Start Break	Ignore Mark Fix Sh	ow again	Auto	
	Settings		Apply	Reset	
	- -				

The wall with very small hole will be marked for later investigation.

Sel.	Correction	Tolerance [m]	Fixed
Х	Delete identical copies		1
Х	Fix overlaps		1
Х	Chamfer sharp angles	0.100	0
Х	Fix small areas and lines	0.100	1
Х	Merge region lines	0.010	0
Х	Align to structure grid	0.100	0
х	Stretch to structure grid	0.100	0
х	Align regions	0.100	0
Х	Stretch to crossing regions	0.100	0
х	Stretch regions in plane	0.100	0
х	Stretch lines	0.100	0
х	Align points	0.100	0
Select	all Clear all		Marker
Start		how again	A vite
Break Settings		Apply	Auto Reset

The next error is the support that is not on Axis 1, but within the given *Tolerance* (0.1 m) for "Align to structure grid". Let's click on **Fix**, to align it to Axis 1.



The number in the *Fixed* column of "Align to structure grid" task is set to 1. The corrected support is green on the next picture. The next misplaced object - the column, above it - is blinking. Let's fix it too.



The number in the *Fixed* column of "Align to structure grid" task is increased to 2. The corrected column is green on the next picture, and the next misplaced object - the large plate - is blinking (since it is slightly above Storey 1). Let's fix it as well and do the same for the next blinking small plate.



Finally, four objects are fixed by "Align to structure grid" task (one support, one column and two plates) and the next incorrect object is the line support that is just a bit short to reach to Axis 1. Let's click **Fix** to stretch it to Axis 1.



The number for the *Fixed* objects in the "Stretch to structure grid" task is set to 1. The next object to be fixed is the wall above the previously fixed line support, which is also too short to reach Axis 1. Let's click **Fix** to stretch it, too.



The next error is found by the "Stretch to crossing regions" task: the plate edge does not fit to the wall below.

orrect mo	odel			8	!	
Sel.	Correction	Tolerance [m]	Fixed			
x	Delete identical copies	[m] 	1			
x	Fix overlaps		1			
x	Chamfer sharp angles	0.100	0			
x	Fix small areas and lines	0.100	1			
х	Merge region lines	0.010	0			
х	Align to structure grid	0.100	4	E		
х	Stretch to structure grid	0.100	2			
x	Align regions	0.100	0			
х	Stretch to crossing regions	0.100	0			
x	Stretch regions in plane	0.100	0			
x	Stretch lines	0.100	0			
x	Align points	0.100	0			
Select	al Gear al		Marker			
Star	t Ignore Mark Fix S	how again				
Brea	k Ignore al Mark al Fix al		Auto			
Setting	p > OK Cancel	Apply	Reset	51		
		T T T	1 1 1			

Then "Stretch lines" task finds a line support that is not connected to the wall above it. Let's click on **Fix** to move it under the wall.

rrect mo	del			8				-
Sel.	Correction	Tolerance [m]	Fixed			<u> </u>		
Х	Delete identical copies		1				1	
х	Fix overlaps		1				1	1.0
х	Chamfer sharp angles	0.100	0				1	
х	Fix small areas and lines	0.100	1			- /-		· · · ·
х	Merge region lines	0.010	0			1	$\sim 1$	1
х	Align to structure grid	0.100	4	E			×12/11	11
х	Stretch to structure grid	0.100	2				T %	
х	Align regions	0.100	0		-/4	- /		·
х	Stretch to crossing regions	0.100	1		1 1	1.00		· · · · · ·
х	Stretch regions in plane	0.100	0	l k		1	12	
Х	Stretch lines	0.100	0					
х	Align points	0.100	0	11		-		
Select	al Cear al		Marker		l			/
Start	Ignore Mark Fix	Show again		=   -			_,i_	
Break	: Ignore al Mark al Fix al		Auto			1		
Settings	Cancel	Apply	Reset					

The next incorrect object is a column that is not exactly above the point support. But we decide that the column is at the right place, so click **Ignore** to skip it.



Now "Align points" task finds the point support which is almost below the column, so let's move it to the bottom of the column by clicking on **Fix**.

Corre	ect mod	el		[
	Sel.	Correction	Tolerance [m]	Fixed
	Х	Delete identical copies		1
	Х	Fix overlaps		1
	Х	Chamfer sharp angles	0.100	0
	Х	Fix small areas and lines	0.100	1
	Х	Merge region lines	0.010	0
	Х	Align to structure grid	0.100	4 ≘
	Х	Stretch to structure grid	0.100	2
	Х	Align regions	0.100	0
	Х	Stretch to crossing regions	0.100	1
- and	Х	Stretch regions in plane	0.100	0
	Х	Stretch lines	0.100	1
· ~	Х	Align points	0.100	0
) 게 E				
Se	ect all	Clear all	(	Marker
	Start	Ignore Mark Fix Sł	now again	
	Break	Ignore all Mark all Fix all	[	Auto
Set	ttings >	OK Cancel	Apply	Reset

Since this is the last error, the green highlighting disappears from the Correction table and **OK** and **Apply** buttons become active.



By clicking on **OK**, the corrections will be applied to the model. The objects that were marked in the dialog are displayed with the marker text.



# 1.2. Scripting

We support a basic automation workflow through scripting. It is capable to load/save file, execute analysis calculation and create outputs as .csv list or .docx documentation. With that you can batch-analyze models created in other programs or directly in .struxml. Or execute long calculations during the night.

The script commands approximate the usual interface, as if you filled inputs on the dialog and press the OK button. So for the meaning of the parameters look at the corresponding panel in the UI.

To launch the script use the menu command Tools/Run script or start the program with the /s command line:

## fd3dstruct /s c:\mydir\example.fdcsript

The installation of FEM-Design 17.01.001 (or later version) creates the **fdscript.xsd** and an **example.fdcsript** in the templates folder, the latter can be used as a starting point of custom scripts. It has all the commands intended for users and useful comments on how to proceed.

Script execution writes a log file (specified within the script header), you have to look into that to see what was actually executed and for any warning/error message that would normally appear on the screen. The execution stops at first serious error.

If you launch FEM-Design many times to execute script for many inputs it's a good idea to turn off all the satellite executables it may launch: update and upgrade in settings and set environment FD\_NOLOGO=1 to avoid the splash screen.

# 1.3. Position numbering tool

All position numbering possibilities are gathered into the new *Tools/Numbering*... tool. It lets the user manually set/reset the object's position numbers, it also contains *Refresh numbering* and *Lock numbering*... commands of FD16.



Four main and two auxiliary options are available in the tool window.



## Manual position numbering

Numbering			
Always ask	n no. conflicts	with other manually set position no.	2
3	B.	.0	
U			

- 1. type required position number into the *Position no*. textbox
- 2. select option for handling position number conflict
- 3. select object(s) to set position number for

In case more objects are selected, the first one gets the position number typed by the user and for the next ones it is increased automatically.

To set position number of **component objects**, like edge connections, corbels, post-tensioned cables, punching regions, the *Select component (...)* auxiliary option has to be checked.

Objects with manually set position number can be **highlighted** by checking the last option of the tool window.

Numbering	Highlight objects with manually set position number
1	Position no 102 If current position no. conflicts with other manually set position no. Always ask
	B.101

#### Reset position number

User can reset the position number to zero by choosing *Reset* option, then selecting one or more objects. *Select component* option is available here, too.

#### Automatic numbering

This option works the same way as *Renumbering* command in FD16. It sets position number automatically for all objects in the database except the ones with manually set position number.

#### Lock numbering

This option works exactly the same way as *Lock numbering*... command in FD16. If it is pressed, the following dialog pops up, where position number of different object types can be locked.

Lock numbering of
<ul> <li>structural objects</li> <li>punching objects</li> <li>steel joints</li> </ul>
OK Cancel

A Manual numbering can not be applied on locked objects.

# 1.4. Section list

A new type of table – *Sections* – can be listed. It contains all sectional data (width, height, area, inertias, etc.)

In the List tables dialog select Tables/ Structures/ Sections, then click on the List table button.

List tables		×
List tables  Tables  Structure  Coads  Coads	Structure         Data         Concrete materials         Steel materials         Steereys         Sections         Beams         Columns         Trusses         Point support groups	X List objects All V Send to Clipboard V Page setup Printer setup Hide irrelevant tables Regional decimal symbol instead of dot Fill all table cells
Max. of combinations		Format table
		Settings > Add to batch
Maximum of load groups	3	List tables

In the generated Section table, the last column – *Other* - contains the detailed dimensions of the section. Currently it is filled only for RHS, I and composite sections.

Sections															
	Section		Compos	ite Heig	ht	Widt	h	A		Р		A/P	•	Yg	Zg
	[-]		[-]	[mm	J I	[mm]		[mm	12]	[mm]	]	[mm]		[mm]	[mm]
Steel section	ons IPE 300	)	No	3	300	1	50	53	381	116	50	4	4.6	0.000	0.000
Steel section	ons IPE 45(	)	No	4	<b>4</b> 50	1	90	- 98	882	160	)5	6	5.2	0.000	0.000
Steel section	ons IPE 33(	)	No	3	30	1	60	62	261	125	54	5	5.0	0.000	0.000
Steel section	ons KKR 40	x20x2	No		40		20	;	214	21	14	1	1.0	0.000	0.000
Steel section	ons KKR 70	x50x2	No		70	1	50		454	49	54	1	1.0	0.000	0.000
Ys	Zs	Iy		Wy	ez n	nax	ez n	nin		iy		Sy		Iz	Wz
[mm]	[mm]	[mm4]	] [	mm3]	[m	m]	[mr	m]	] [mm]		[mm3]		[mm4]		[mm3]
0.000	0.000	8356113	32	557074		150		150	50 124.6		3	314179		6037784	80504
0.000	0.000	33742960	33	1499687		225	225		5 184.8		8	850900		16758611	176406
0.000	0.000	11766909	93	713146		165		165		137.1	4	02166		7881421	98518
0.000	0.000	404	95	2025		20		20		13.8		1307		13428	1343
0.000	0.000	3147	55	8993		35		35		26.3		5400		187577	7503
ey max	ey min	iz	Sz	I	t		Wt			Iw		Iyz		z omega	alpha1
[mm]	[mm]	[mm]	[mm3]	[mr	n4]	[n	nm3]		[r	mm6]		[mm4]	]	[-]	[°]
75	75	33.5	6264	4 19	7538		1131	18 1	12425	6407470	)		-0	0	0.000
95	95	41.2	138282	66	0584		2737	70 7	78096	6316453			0	0	0.000
80	80	35.5	7688	2 27	5905		1397	'3  1	19608	8467329			-0	0	0.000
10	10	7.9	79	в з	4618		219	2		28780)	7		0	0	0.000
25	25	20.3	429	L 37	5319		964	¥1		320669	1		0	0	0.000

I1	W1 min	W1 max	e2 max	e2 min	i1	S1	S01	c1	Rho 1
[mm4]	[mm3]	[mm3]	[mm]	[mm]	[mm]	[mm3]	[mm3]	[-]	[-]
83561132	557074	557074	150	150	124.6	314179	314179	1.128	0.546
337429603	1499687	1499687	225	225	184.8	850900	850900	1.135	0.519
117669093	713146	713146	165	165	137.1	402166	402165	1.128	0.546
40495	2025	2025	20	20	13.8	1307	1307	1.291	0.230
314755	8993	8993	35	35	26.3	5400	5400	1.201	0.326

[	z2	alpha2	I2	W2 min	W2 max	e1 max	e1 min	i2	S2	S02
[	[mm]	[°]	[mm4]	[mm3]	[mm3]	[mm]	[mm]	[mm]	[mm3]	[mm3]
ſ	0	1.571	6037784	80504	80504	75	75	33.5	62644	62644
	0	1.571	16758611	176406	176406	95	95	41.2	138282	138279
	0	1.571	7881421	98518	98518	80	80	35.5	76882	76882
	0	1.571	13428	1343	1343	10	10	7.9	798	798
	0	1.571	187577	7503	7503	25	25	20.3	4291	4291

c2	Rho 2	z1	Other
[-]	[-]	[mm]	[-]
1.556	0.386	0	tw=7.1mm; hw=279mm; tf=10.7mm; wf=150mm; r=15.0mm
1.568	0.419	0	tw=9.4mm; hw=421mm; tf=14.6mm; wf=190mm; r=21.0mm
1.561	0.388	0	tw=7.5mm; hw=307mm; tf=11.5mm; wf=160mm; r=18.0mm
1.189	0.659	0	t=2.0mm; r=2.00mm
1.144	0.536	0	t=2.0mm; r=2.00mm
-			

## 1.5. Load combination analysis setup list

In the *Calculation/Setup by load combination* dialog, a new option called *Add to documentation* is available to document the analysis setup for load combinations.

lo	Туре	Load combination	NLE	PL	NLS	Cr.	2nd	Im.	Ground water level	1
1	U	1.35*1 + 1.50*3 + 1.50*0.70*4	х	Х						
2	U	1.35*1 + 1.50*0.70*3 + 1.50*4	X	Х						
3	Sq	1+0.20*3+0.20*4	X	Х						
4	Sf	1+0.50*3+0.20*4	X	Х						
5	Sf	1+0.20*3+0.50*4	X	Х						
6	Sc	1+3+0.70*4	X	Х						
7	Sc	1 + 0.70*3 + 4	X	x						

The table in the documentation:

Load	combi	nations analysis setup							
No	Туре	Load combination	Non-linear	Plastic	Non-linear	Cracked	2nd	Imperfection	Ground
			elements	elements	soil	section	order	shape	water level
1	υ	1.35*1 + 1.50*3 + 1	Yes	Yes	No	No	No	No	-
2	υ		Yes	Yes	No	No	No	No	-
3	Sq	1 + 0.20*3 + 0.20*4	Yes	Yes	No	No	No	No	-
4	SF	1 + 0.50*3 + 0.20*4	Yes	Yes	No	No	No	No	-
5	SF	1 + 0.20*3 + 0.50*4	Yes	Yes	No	No	No	No	-
6	Sc	1 + 3 + 0.70*4	Yes	Yes	No	No	No	No	-
7	Sc	1 + 0.70*3 + 4	Yes	Yes	No	No	No	No	-

This option can be reached from the List tables dialog, too. Pick Tables/Loads/Load combinations.



# 1.6. Arranging tables when listing to Excel

New options have been implemented into the feature *List to Excel*. From now on User can choose, if the tables should be placed into different Excel-worksheets or into the same worksheet under or next to each other.

List to Excel	x
File path	
	>
Template path	
	>
Open file in Excel	
All listed tables (not included in the template) are placed on	
🔘 a different worksheet	
It the same worksheet under each other	
the same worksheet next to each other	
OK Cancel	

# 1.7. "Fill all cells" option for listing tables

In order to ease sorting data in Excel after exporting tables, User can fill the empty cells by turning on the *"Fill all table cells"* option. This option is also available in the Documentation, in the *Table properties* dialog in the *Option* tab.

List objects	Table properties
Current selection	Format Options
Send to Clipboard	List objects
Page setup	Current selection 🔻
Printer setup	Fill all table cells
Hide irrelevant	Results of surface elements
Regional decimal symbol instead of dot	<ul> <li>In center of element</li> <li>In vertices</li> </ul>
Fill all table cells	

The left part of table on figure below shows the result if the checkbox is off, and right part shows when it is on.

30	•	:	$\times$	✓ fx														
A	в	с	D	E	F	G	н	I	J	K	L	м	N	0	Р	Q	R	S
Shells	, Displa	cement	ts, Ultim	ate - Load	case: f - f	or selecte	d object	s		Shells, I	Displacen	nents, Ulti	mate - Lo	oad case:	f - for sele	ected obj	ects	
Shell	Elem N	lode	ex'	ey'	ez'	fix'	fiy'	fiz'		Shell	Elem	Node	ex'	ey'	ez'	fix'	fiy'	fiz'
[-]	[-] [·	-]	[mm]	[mm]	[mm]	[rad]	[rad]	[rad]		[-]	[-]	[-]	[mm]	[mm]	[mm]	[rad]	[rad]	[rad]
P.1.1	1	479	0.000	0.000	-8.386	0.006	-0.002	0.000		P.1.1	1 :	1 479	0.000	0.000	-8.386	0.006	-0.002	0.000
	2	538	0.000	0.000	-1.272	0.002	-0.001	0.000		P.1.1	1 2	2 538	0.000	0.000	-1.272	0.002	-0.001	0.000
	3	640	0.000	0.000	-2.653	-0.000	-0.004	0.000		P.1.1	1 3	3 640	0.000	0.000	-2.653	-0.000	-0.004	0.000
	4	234	0.000	0.000	-36.524	0.015	-0.001	0.000		P.1.1	4	4 234	0.000	0.000	-36.524	0.015	-0.001	0.000
	5	136	0.000	0.000	-88.207	0.004	-0.002	0.000		P.1.1		5 136	0.000	0.000	-88.207	0.004	-0.002	0.000
	6	591	0.000	0.000	-11.409	0.000	-0.009	0.000		P.1.1	(	5 591	0.000	0.000	-11.409	0.000	-0.009	0.000
	7	347	0.000	0.000	-70.238	-0.003	-0.005	0.000		P.1.1		7 347	0.000	0.000	-70.238	-0.003	-0.005	0.000
	8	632	0.000	0.000	-0.594	-0.000	-0.001	0.000		P.1.1	8	3 632	0.000	0.000	-0.594	-0.000	-0.001	0.000
	9	579	0.000	0.000	-0.594	-0.001	-0.000	0.000		P.1.1		579	0.000	0.000	-0.594	-0.001	-0.000	0.000
	10	598	0.000	0.000	-2.542	0.002	-0.002	0.000		P.1.1	10	598	0.000	0.000	-2.542	0.002	-0.002	0.000
	11	288	0.000	0.000	-34.532	0.014	-0.002	0.000		P.1.1	1:	1 288	0.000	0.000	-34.532	0.014	-0.002	0.000
	12	526	0.000	0.000	-11.566	0.005	-0.005	0.000		P.1.1	12	2 526	0.000	0.000	-11.566	0.005	-0.005	0.000
	13	394	0.000	0.000	-28.245	0.012	-0.003	0.000		P.1.1	13	3 394	0.000	0.000	-28.245	0.012	-0.003	0.000
	14	286	0.000	0.000	-68.462	0.009	-0.003	0.000		P.1.1	14	4 286	0.000	0.000	-68.462	0.009	-0.003	0.000
	A Shells Shell [-]	A         B           Shells, Displa         Shell           Shell         Elem         N           [-1]         [-1]         [           P.1.1         1         2           3         4         5           6         7         6           7         8         9           10         11         12           12         13         13	A         B         C           Shells, Displacement         Shell Elem Node           Shell         [-]         [-]           P.1.1         1         479           2         538         3           4         234         5           5         136         6           6         591         7           8         632         9         579           10         598         11         288           11         2         256         13           13         394         13         394	A         B         C         D           Shells, Displacements, Ultim         Node         ex'           Shell Elem         Node         ex'           [-]         [-]         [-]         [mm]           P.1.1         1         479         0.000           2         538         0.000           4         234         0.000           4         234         0.000           5         136         0.000           6         591         0.000           7         347         0.000           9         579         0.000           10         598         0.000           11         288         0.000           12         2526         0.000           13         394         0.000	A         B         C         D         E           Shells, Displacements, Ultimate - Load         Shells, Displacements, Ultimate - Load         Shells, Displacements, Ultimate - Load           Shell Elem         Node         ex'         ey'         I           [-]         [-]         [-]         [mm]         [mm]           P.1.1         1         479         0.000         0.000           3         640         0.000         0.000           4         234         0.000         0.000           5         136         0.000         0.000           6         591         0.000         0.000           7         347         0.000         0.000           9         579         0.000         0.000           10         598         0.000         0.000           11         288         0.000         0.000           12         526         0.000         0.000           13         394         0.000         0.000	A         B         C         D         E         F           Shells, Displacements, Ultimate - Load case: f - for Shell Elem Node         ex'         ey'         ez'         ez'           [-1]         [-1]         [-1]         [mm]         [mm]         [mm]         [mm]           P.1.1         1         479         0.000         0.000         -8.386           2         538         0.000         0.000         -2.653           4         234         0.000         0.000         -36.524           5         136         0.000         0.000         -11.409           7         347         0.000         0.000         -0.594           9         579         0.000         0.000         -2.542           10         588         0.000         0.000         -2.542           11         288         0.000         0.000         -34.532           12         526         0.000         0.000         -34.532           13         394         0.000         0.000         -28.245	A         B         C         D         E         F         G           Shells, Displacements, Ultimate - Load case: f - for selecte         Shells, Displacements, Ultimate - Load case: f - for selecte         Shells, Displacements, Ultimate - Load case: f - for selecte           Shell Elem         Node         ex'         ey'         ez'         fix'           [-1]         [-1]         [-1]         [mm]         [mm]         [mm]         [rad]           P.1.1         1         479         0.000         0.000         -8.386         0.006           2         538         0.000         0.000         -2.653         -0.000           4         234         0.000         0.000         -36.524         0.015           5         136         0.000         0.000         -70.238         -0.003           8         632         0.000         0.000         -0.594         -0.001           10         598         0.000         0.000         -2.542         0.002           11         286         0.000         0.000         -34.532         0.014           12         526         0.000         -1.566         0.005           11         286         0.000         0	A         B         C         D         E         F         G         H           Shells, Displacements, Ultimate - Load case; f - for selected object         Shells, Displacements, Ultimate - Load case; f - for selected object           Shell Elem         Node         ex'         ey'         ez'         fix'         fiy'           [-1]         [-1]         [mm]         [mm]         [mm]         [rad]         [rad]           P.1.1         1         479<0.000	A         B         C         D         E         F         G         H         I           Shells, Displacements, Ultimate - Load case: f - for selected objects           Shell Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'           [-1]         [-1]         [-1]         [mm]         [mm]         [mm]         [rad]         [rad]         [rad]           P.1.1         1         479         0.000         0.000         -8.386         0.006         -0.002         0.000           2         538         0.000         0.000         -2.653         -0.000         0.000           3         640         0.000         -2.653         -0.001         0.000           4         234         0.000         0.000         -36.524         0.015         -0.001         0.000           5         136         0.000         0.000         -70.238         -0.003         -0.002         0.000           6         591         0.000         0.000         -5.54         -0.01         -0.002         0.000           8         632         0.000         0.000         -5.54         -0.003         -0.002         0.	A         B         C         D         E         F         G         H         I         J           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects           Shell Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'           P.1.1         1         479         0.000         0.000         -8.386         0.006         -0.002         0.000           2         538         0.000         0.000         -1.272         0.002         -0.001         0.000           4         234         0.000         0.000         -36.524         0.015         -0.001         0.000           5         136         0.000         0.000         -11.499         0.000         -0.001         0.000           6         591         0.000         0.000         -0.032         -0.000         0.000           7         347         0.000         0.000         -0.031         -0.000         0.000           9         579         0.000         0.000	A         B         C         D         E         F         G         H         I         J         K           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells,         Shells,<	A         B         C         D         E         F         G         H         I         J         K         L           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacem         Shells, Displacem         Shells, Displacem           Shell         Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'         Shell         Shell         Elem           [-1]         [-1]         [-1]         mm]         [mm]         [mm]         [rad]         [rad]         [-1]<	A         B         C         D         E         F         G         H         I         J         K         L         M           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultim           Shell Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'         Shells, Displacements, Ultim           P.1.1         1         479         0.000         0.000         -8.386         0.006         -0.002         0.000         P.1.1         1         479           P.1.1         1         479         0.000         0.000         -8.386         0.000         0.000         P.1.1         1         479           P.1.1         1         479         0.000         0.000         -2.653         -0.001         0.000         P.1.1         2         538           3         640         0.000         -2.653         -0.001         0.000         P.1.1         4         234           5         136         0.000         -36.524         0.015         -0.001         0.000         P.1.1         5         136           6	A         B         C         D         E         F         G         H         I         J         K         L         M         N           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects           P.1.1         1         479         0.000         .000         -1.1         I         479         0.000           2         538         0.000         .0001         .0000         P.1.1         I         479         0.000           3         640         0.000         .0001         .0001         .0000         P.1.1         4         234         0.000           5         136         0.000         .0000	A         B         C         D         E         F         G         H         I         J         K         L         M         N         O           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case:         Shells, Displacement	A         B         C         D         E         F         G         H         I         J         K         L         M         NO         P           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects           P.1.1         1         479         0.000         0.000         -8.386         0.000         P.1.1         1         479         0.000         0.000         -2.653           2         538         0.000         0.000         -2.653         0.001         0.000         P.1.1         4         234         0.000         0.000	A         B         C         D         E         F         G         H         I         J         K         L         M         O         P         Q           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shell Elem         Node         ex'         ey'         ez'         fix'         fix'         Shell Elem         Node         ex'         ey'         ez'         fix'         fix'         Shell Elem         Node         ex'         ey'         ez'         fix'         fix'         fix'         Shell Elem         Node         ex'         ey'         ez'         fix'         fix' <t< td=""><td>A         B         C         D         E         F         G         H         I         J         K         L         M         N         O         P         Q         R           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects           Shell         Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'         Shells, Displacements, Ultimate - Load case: f - for selected objects           Shell         Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'         Shell         Film         Node         ex'         ez'         fix'         fiy'         fiz'         Shell         Film         Node         ex'         ez'         fix'         fiy'         fiz'         Shell         Film         1         479         0.000         0.000         -0.001         Node         ex'         ez'         fix'         fiy'         fiz'         fix'         fiz'         Shell         Shell         Shells         Displacements, Ultimate - Load case: f - for sel</td></t<>	A         B         C         D         E         F         G         H         I         J         K         L         M         N         O         P         Q         R           Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects         Shells, Displacements, Ultimate - Load case: f - for selected objects           Shell         Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'         Shells, Displacements, Ultimate - Load case: f - for selected objects           Shell         Elem         Node         ex'         ey'         ez'         fix'         fiy'         fiz'         Shell         Film         Node         ex'         ez'         fix'         fiy'         fiz'         Shell         Film         Node         ex'         ez'         fix'         fiy'         fiz'         Shell         Film         1         479         0.000         0.000         -0.001         Node         ex'         ez'         fix'         fiy'         fiz'         fix'         fiz'         Shell         Shell         Shells         Displacements, Ultimate - Load case: f - for sel

# **1.8.** Displaying name of load case/combination for load case/combination result tables

In FD 17 the name of the load combinations and load cases can be displayed for the load case and load combination result tables.

Check Show hidden columns checkbox in Table properties dialog:

Table properties	×
Format Options	1.
AaBbYyZz •	
abcdefg abcdefg 123 123.457 12	d] .457 .457
Bocdefg         123         123.457         12	(437)
✓ Hide           Width [mm]           15	
Scientific (exponent) form         Decimals	
	/ hidden columns
OK	Cancel



Shell	Elem	Node	ex'	ey'	ez'	fix'	fiy'	fiz'	Case
[-]	[-]	[-]	[mm]	[mm]	[mm]	[rad]	[rad]	[rad]	[-]
P.1.1	1	479	0.000	0.000	-8.386	0.006	-0.002	0.000	f
P.1.1	2	538	0.000	0.000	-1.272	0.002	-0.001	0.000	f
P.1.1	3	640	0.000	0.000	-2.653	-0.000	-0.004	0.000	f
P.1.1	4	234	0.000	0.000	-36.524	0.015	-0.001	0.000	f
P.1.1	5	136	0.000	0.000	-88.207	0.004	-0.002	0.000	f
P.1.1	6	591	0.000	0.000	-11.409	0.000	-0.009	0.000	f
P.1.1	7	347	0.000	0.000	-70.238	-0.003	-0.005	0.000	f
D 1 1	0	622	0.000	0.000	0 504	0.000	0.001	0.000	f

# 1.9. Get GUID (Globally Unique Identifier)

The *Get GUID* function enables the query of the GUID of elements. This can be useful for identification of structural objects imported/exported via Struxml.

The function-button can be found in the *Tools* menu, but it is not on the *Toolbar* ( $\bigcirc$ ) by default.User can put it there by using *Customize*... command which is available by right clicking on the toolbar.



After launching *Get GUID*, pick the element(s). The pop-up window shows the Globally Unique Identifiers of selected element(s). They can be sent to the clipboard by clicking on "Send to clipboard" button. GUIDs for analytical and physical model of a bar are different.



If the bar is saved in struxml format, these GUIDs can be found there as well.



# 2. User interface

## 2.1. Properties dialog available from Quick menu for objects

Simply select any object by right click, or more objects of the same type by box selection, then click *"Properties*" in the Quick menu to check/modify its/their properties. This function works for structural objects, loads or design elements.



Objects that are part of other object (e.g. edge connections, corbels, reinforcement regions, etc.)usually can be selected by left-to-right (blue) box selection only, because right clicking always selects the main object (shell, bar).



# 2.2. Different modes to display storeys

In *Settings/Display/Storey* dialog User can choose from three options for the appearance of the given storey.



All objects that are in the plane of the storey, or crossing the storey above or below according to the selected setting, are displayed

The pictures below show the whole structure and how it is displayed according to the selected option.


# 2.3. Visualisation of edge connection is improved

In the new version, *Edge connections* are displayed with customizable lanes.



The width of the edge connection can be modified in the Settings/Display/Shells dialog.

Structural element ID       Display local system         Analytical element ID       Size [m]         Panel type ID       Display edge connection         Eccentricity       Width [m]         Material       Hatch [mm]         E2 / E1       Display wall base line         Alpha (orthotropic angle)       Thickness	isplay label including	Graphical options
□ Panel type ID       ☑ Display edge connection         □ Eccentricity       Width [m]         □ Material       Hatch [mm]         □ E2 / E1       □ Display wall base line	Structural element ID	Display local system
Image: Second trial for the second trine second trine second trial for the second trial for the second t	Analytical element ID	
Material     Hatch [mm]     1.5000       E2 / E1     Display wall base line       Alpha (orthotropic angle)     Image: Comparison of the second secon	Panel type ID	
E2 / E1 Display wall base line	Eccentricity	Width [m] 0.300
Alpha (orthotropic angle)	Material	Hatch [mm] 1.5000
	E2/E1	Display wall base line
	Alpha (orthotropic angle)	
	Thickness	

## 2.4. Restore view returning to input

In	put tabs			An	alysis and de	esign tabs		
File Edit	Draw Modify Too	ls Settings V	iew Window	Help				
Structure	Loads Finite eleme	nts Analysis	Foundation d	esign RC desig	gn Steel design	Timber design	Performance based de	sign
•	• 1 A			√ 🔁			ri (ri (ri	4
<	Structure grid	▶ Fo	Bar			Shell		

Once a view set in an input tab gets modified in the Analysis or in a design tab, User can return into an input tab while automatically restoring the original view. See an example in the following figures.

Using Ctrl:



#### Back to original view





Without pressing Ctrl the view doesn't switch back:

#### Back to original view



#### 2.5. Leading line for numeric values and labels

If a numeric value or a label is moved away from the referred point, a thin leading line appears.



## 2.6. Align region to plane

This function will align regions to selected plane in any direction with two clicks. The feature can be found in *Modify/Modify region*.



## 2.7. Vertical dimension lines

From now vertical dimension lines can be placed <u>without modifying the user coordinate-system</u>. In the *Dimensioning dialog* select the parallel direction with global Z axis or with z axis of UCS.



## 2.8. Physical view

In FEM-Design 17 the analytical and physical view of an object is clearly separated. The physical view shows the real position of the physical element, while the analytical view shows the calculation model.



## 2.9. Physical alignment for intermediate section

In version 17 the User has the option to set the physical and analytical eccentricity separately for intermediate section.



#### 2.10. Blinking layers

In FEM-Design 17, upon right clicking on any Layer in the *Layers* dialog, all elements belonging to that Layer will blink. This works both for Drawing and Objects layers.

Additionally, a new Layer called Import error is added into Drawing layers. This is relevant only in case when the model was imported through struxml and contained some bad shell objects. In case those shell objects could not be imported, they will be shown (in some cases) as graphical image and placed on above mentioned layer.



# 3. Structure

## 3.1. Improved storey dialog

If the *Modify structure* is enabled the structure follows the storey system modification, otherwise not. The modifications are displayed with three color code, if the *Modify structure* is active:

- green: new storey
- yellow: modified storey
- red: deleted storey

When the *Apply* button is clicked the modifications proceed immediately without closing the dialog. The following pictures shows this feature though a 4-storey building, where the *Storey 4* will be **deleted**, a **new storey is added** and the *Storey 3* **height is increased** to 4.0 meters.



#### 3.2. Reference plane

Reference plane is an auxiliary object, which allows User to align elements to it. It can be found in the Structure tab.



As is it shown below, *Description* of *Reference plane* can be typed in the tool palette or after clicking on the Default properties button, it can be set in the new dialog.

Reference plane	Reference plane
P   C     Description   User defined name	Description User defined name
	OK Cancel

*Reference plane* can be drawn in any plane. It is displayed by the region contours and its description given by the user.



The *Reference plane* can be used with the feature *Align region to plane*. (Modify/Modify region/Align region to plane/Pick region to align/Select *Reference plane* - See chapter 2.6) and it also can be used by *Correct model* tool to fit structural objects to.



*Reference planes* can be exported into/imported from Autodesk Revit via struxml format.

## 3.3. Composite sections

/!\

The composite sections below have been implemented for *Beams* and *Columns*. These sections can be reached in *Beam/Column dialog/Section tab/Composite/New*.



For the time being only above listed composite sections can be selected, there is no choice to create alternative section.

## 3.4. Pile

In FEM-Design 17, a new structural element is added - the pile. Its main purpose is to calculate pile internal forces in either linear or non-linear way (e.g. limited shaft friction, only compression behaviour at the tip of pile can be considered).

The basic concept is that the soil surrounding the pile is modelled with continuous line supports along it. These supports stand for the supporting effect of the soil together with a vertical point support at the tip of pile. The new plastic element feature offers a more precise way to calculate the internal forces by considering the nonlinear characteristic of the soil-structure interaction.

This feature can be found among the Foundation objects on Structure tab.



Piles can be placed either one by one, or in groups using the "Array rectangular"/ "Array polar" options. In the Tool window User can declare the *Height* of the pile and set the array options as *Column distance*, *Row distance* and *Angle of array*.

Pile		Array rectangular	x
	↑ †I <u>↓</u>  +‡	Array polar	
	Height [m] 8.0000	Column distance [m] 3.0000	
TTT	1-	Row distance [m] 3,0000	
		Angle of array [°] 30.000	
	y' axis is parall y' axis is parall y' axis is parall y' axis is parall	el with XY plane el with YZ plane el with XZ plane el with UCS	

If choosing *Array rectangular* define option, at first User needs to set the *Column* and *Row distances*, before placing the piles. By clicking these textboxes turn active.



If choosing Array polar, two more options become available according to the following figure:



To open Pile dialog click Default settings.

Pile			<b>E</b>
	↑ †  <u>↑</u>   •‡ 💽 I (		
	Height [m] 8.0000	Column distance [m]	3.0000
TTT	1-	Row distance [m]	3.0000
L Default se	ttings	Angle of array [°]	30.000

On General tab User can give an identifier and a rotation angle.

Pile	×	
	A.1 General Section Aterial End conditions	
	Identifier (.position number) PI	
	Rotation	
	Alpha [°]0.000 Alpha is measured from default direction of y' axis. The default direction is usually horizontal and in case of vertical elements points to global y direction.	

Regular and certain types of composite sections can be selected in Section tab/Composite/ New.

Pile	Pile
A.1 General Section A Material End conditions	A.1 General Section Material End conditions
Section type Regular Composite	Section type Regular Composite
The same at both ends Start End	Composite sections Ultimate limit state
Section library       ▲	New Modify Delete
OK Cancel	OK Cancel

The available composite sections for simple pile are the followings:



The material can be selected in the same way as for *Beams/Columns*.

		Material End conditions Soil springs
Library	*	Application data
Concrete		
		Gamma c (U   Ua,Us) 1.50 1.20
C16/20 C20/25		Gamma s (U   Ua,Us) 1.15 1.00
C25/30		
C30/37	E	Gamma cE 1.20 >
C35/45		Alpha cc
C40/50		
C45/55		Alpha ct 1.00
C50/60		Creep c. (S   U) 0.000 0.000
⊡ Steel		
S 235 S 275		Shrinkage [‰]
S 355		Reduction for dynamic analysis 1.00
- S 420		Reduction for stability analysis 1.00
S 450		Reduction for stability analysis 1.00
S 460		
- Timber		
C14		
C20 C22		
C24		
	*	
New Modify	Delete	Import Export

On the *End conditions* tab User can set the top release of the pile, which can be useful in case piles are connected to a foundation slab. The connection can be either fixed or hinged.



The program generates a set of line support groups for the pile (at least one for each stratum with an additional breakpoint at the highest water level) and one point support at the tip of the pile. If the pile is edited or some changes happen in the input data (e.g. changing of the material of a stratum), these stiffness and plastic limit values will be recalculated. The automatic calculation can be switched off in *Pile dialog/Soil springs tab/Auto calculate checkbox*.

Pile			×
A.1 General S	ection 🚺 Material	End conditions	Soil springs
Support springs Ne	g. shaft friction rder)	/	
Lateral line motion spring	- D-M (m (m)	Latera plastic limit force	- RA1(m]
Compression	Tension	Compression	Tension
Ky'	×		
Kz' 📉	~		
Vertical line motion spring	s [kN/m/m]	Vertical plastic limit force	es [kN/m]
Compression	Tension	Compression	Tension
Kx'			
Vertical pile tip spring [kN	/m]	Vertical plastic limit force	es [kN]
Compression	Tension	Compression	Tension
Kx'			
Options Auto	calculate		
			OK Cancel
			Calicel



The *Soil springs tab* will be active only after placing the pile and its properties is displayed in the dialog.

From the previously defined strata, FEM-Design calculates the stiffness and plastic limit values for both the line and point supports.

The *line support groups* along the pile have only translational stiffness,  $K_{x'}$  vertical (shear) and  $K_{y'}$ ,  $K_{z'}$  horizontal stiffness [kN/m/m]. For the *horizontal spring stiffness*, it does not have any practical sense to define compression and tension stiffness, thus for the horizontal directions (y', z') one value defines each direction. The horizontal stiffness of the line support in y' direction (similar in z' direction):

$$K_{y}' = k_{s,y'} \cdot B \quad \left[\frac{kN}{m^2}\right]$$

where B is the width of the pile and ks,y' is the horizontal coefficient of subgrade reaction. According to *Vesic (1961),* it can be calculated by using both soil and pile properties:

$$k_{s,y'} = \frac{0.65 \cdot E_s}{B \cdot (1 - \mu_s^2)} \cdot \left[\frac{E_s \cdot B^4}{E_p \cdot I_{p,z'}}\right]^{\frac{1}{12}} \quad \left[\frac{kN}{m^3}\right]^{\frac{1}{12}}$$

where  $E_s$  and  $\mu_s$  is the Young's modulus and Poisson's ratio of the soil,  $E_p$  and  $I_{p,z'}$  are the Young's modulus and moment of inertia of the pile.

The vertical behaviour, thus the *vertical spring stiffness* might be different for compression and tension, so both spring stiffness are available for the User to define (for example One may want to neglect the friction forces for tensioned piles). Based on the analytical solution of *Zhang Q. et al. (2014)*, the vertical stiffness of the line support is:

$$K_{x'} = k_s \cdot P = \frac{G_s}{r_0 \cdot \ln\left(\frac{r_m}{r_0}\right)} \cdot P \quad \left[\frac{kN}{m^2}\right]$$

where  $G_s$  is the shear modulus of the soil,  $r_0$  is the pile radius (or equivalent radius for noncircular piles),  $r_m$  is the radial distance at which shear stresses in the soil become negligible and P is the perimeter of the cross section of the pile. In the different soil layers,  $r_m$  distance can be calculated as the following:

$$r_m = 2.5 \cdot L \cdot \rho_m \cdot (1 - \mu_s)$$

where L is the (total) length of the pile,  $\rho_m$  is the factor of vertical homogeneity of soil stiffness and  $\mu s$  is the Poisson's ratio of soil layer around the pile. The value of  $\rho_m$  can be calculated in the following way:

$$\rho_m = \frac{G_{s,middle}}{G_{s,bottom}}$$

where  $G_{s,middle}$  is the shear modulus at the middle of the layer and  $G_{s,bottom}$  is the shear modulus at the bottom of the layer (relevant only in case of soils with variable material properties with height).

The point support at the tip of the pile has only vertical (or in case of skew piles only x' directional) stiffness. The tension stiffness is set to zero by default as tension forces are not transmitted between the soil and the tip of the pile. *Zhang Q. et al. (2014)* suggest the following formulae to use for the compression stiffness:

$$K_{x'} = \frac{4 \cdot G_s \cdot r_0}{(1 - \mu_s)} \left[\frac{kN}{m}\right]$$

Besides the spring stiffness, plastic limit values are also calculated for both line and point supports. For the vertical line support the plastic limit value is based on the shaft friction, for the point support at the tip of the pile it depends on the end bearing of the pile. In both cases the calculation is affected by the behaviour of the soil (drained/undrained), as it is summarized by *Wrana B (2015)*.

In case of *undrained soil* the limit value is based on the undrained shear strength (c<sub>uk</sub>). The limit value of the *line support*:

$$P_{lim,x'} = q_s \cdot P = \alpha \cdot c_{uk} \cdot P \quad \left[\frac{kN}{m}\right]$$

where  $q_s$  is the shaft friction and  $\alpha$  is the adhesion coefficient. The latter one is calculated according to the proposition of the *NAVFAC DM 7.2 (1984)*, the corresponding details can be found in the literature. The limit value of the *point support* (for compression):

$$P_{lim,x'} = A_{base} \cdot c_{uk} \cdot N_c \ [kN]$$

where  $A_{\text{base}}$  is the cross-sectional area of the pile (at the tip),  $N_c$  is the bearing capacity coefficient that can be assumed equal 9.0 according to *Skempton (1959)*.

In case of *drained soil* the resistances are based on the internal friction angle ( $\varphi k$ ) and cohesion (*ck*) of the soil layers. The limit value of the *line support*:

$$P_{lim,x'} = q_s \cdot P = \beta \cdot \sigma'_v \cdot P \quad \left[\frac{kN}{m}\right]$$

where  $\sigma_{v'}$  is the vertical effective stress and  $\beta$  is a friction coefficient multiplier. As the vertical stresses are increases with the depth, an average value of the current layer is used in the equation. At water level vertical stresses have a break point, thus at the generation of line supports this point is always considered in such a way that it is the start/endpoint of the neighbouring line supports.



Only the *highest water level* is considered in the calculation of plastic limit values and negative shaft friction forces.

The value of  $\beta$  is taken according to the *NAVFAC DM 7.2 (1984)*. The limit value of the *point support* (for compression):

$$P_{lim,x'} = A_{base} \cdot q_b = A_{base} \cdot \left(\sigma'_{v} \cdot N_q + c_k \cdot N_c\right) [kN]$$

Where

$$N_c = \left(N_q - 1\right) \cdot \cot \phi_k$$

where Nq is the bearing capacity factor, and according to the NAVFAC DM 7.2 (1984), its value depends on the fabrication technology of the pile and the internal friction angle of the soil. In the Support springs side of the Soil springs tab User can overwrite the automatically calculated values of any support, such as Lateral/Vertical line motion springs, Vertical pile tip springs or Vertical plastic limit forces.

ile			>	
A.1 General Section Aterial End conditions Soil springs				
Line supports (top down order) Material	No	Ks,y' Rol(m3)	Ks,z'	
soil 1	1	5116	5720	
	2	5116	5720	
	3	5116	5720	
	4	5116	5720 🗸	
Lateral line motion springs [kV/m/m] Compression Tension Ky' 767 $\checkmark$ 767 Kz' 686 $\checkmark$ 686	~	Lateral plastic limit for Compression 1.000e+15 1.000e+15	rces [kN/m] Tension 1.000e+15 1.000e+15	
Vertical line motion springs [kN/m/m] Compression Tension		Vertical plastic limit fo	orces [kN/m] Tension	
Kx' 1071 V V 1071	~	☑ 30.0	30.0	
Vertical pile tip spring [kN/m] Compression Tension Kx' 276 V + 0.000e+4	00 ~	Vertical plastic limit fo Compression 21.1	Tension	
Options Auto calculate				
		[	OK Cancel	

The internal forces of the compressed piles should be increased by the value of the negative shaft friction, as it is an additional effect coming from the soil-pile interaction. In FEM-Design it is considered with a special, automatically generated load case. This load case exits only if there is at least one pile in the model. As negative shaft friction appears only above the *neutral level* (measured from the top of the pile), by default it is set to zero, which means that there are no actual loads in this load case. After we change the neutral level, the corresponding loads are calculated and generated automatically.

The value of the friction force in case of undrained soil:

$$P_s = P \cdot \alpha_{neg} \cdot c_{uk} \ [\frac{kN}{m}]$$

where  $\alpha_{neg}$  is the factor for negative shaft friction for undrained soils (default value is assumed to 1.0). The friction force for *drained soil*:

$$P_s = P \cdot \beta_{neg} \cdot \sigma'_{\nu} \quad [\frac{kN}{m}]$$

where  $\beta_{neg}$  is the factor for negative shaft friction for drained soils (default value is also assumed to 1.0). This load is applied as a linearly variable line load along the pile, and its value can be changed by User by the modification of the  $\alpha_{neg}$  and  $\beta_{neg}$  values (each stratum has one value) together with neutral level on the *Neg. shaft friction* side of the *Soil springs tab*:



If User clicks on *Option*, the *Pile option dialog* opens. Here User can select the *Type of pile*, declare the *Surface surcharge* (affecting the vertical stresses in case of drained soils), the *division length of line supports* and can decide the method of the *Section perimeter's* calculation. Division length becomes a very important setting in case on plastic analyses. The plastic limit forces are increasing with depth, but along a line support it is constant. Consequently, the more line supports are created along the pile (with the automatically calculated, increasing limit values) the more accurate plastic distribution of internal forces we get.





The feature has some important limitations!

- Pile model is used only for analysis purpose (displacements, internal forces), geotechnical design calculations are not implemented yet.
- During the calculations soil must not be modelled as soil element. Soil



- Negative shaft friction is considered in a load combination only if it includes "Neg. shaft friction" load case, which is generated automatically when a pile is created in the model with non-zero neutral level.

Load cas	es					×
No	Name	Туре	Duration (EN 199		ОК	
1 Ne	g. Shaft friction	Neg. Shaft friction	Permanent		Cancel	
inations	<u>++</u>	Load case, combi		<b>↓</b> A↓	<b>↓</b> +↓	×
	Name		Туре	Factor	Included load cases	ОК
shaft friction			U		Neg. Shaft friction 🔹	Cancel
					Neg. Shaft friction Seismic max. Seis res, Fx+Mx Seis res, Fx-Mx Seis res, Fy+My Seis res, Fy-My	Import / Export >

References:

Load

No

- Bogumił Wrana (2015) Pile load capacity – calculation methods. Studia Geotechnica et Mechanica, Vol. 37, No. 4, pp. 83-93

Seis res, Fz

- NAVFAC DM 7.2 (1984): Foundation and Earth Structures, U.S. Department of the Navy
- Skempton A.W. (1959), Cast-in-situ bored piles in London clay, Geotechnique, Vol. 9, No. 4, pp. 153–173
- Qian-qing Zhang, Shu-cai Li, Fa-yun Liang, Min Yang, Qian Zhang (2014) Simplified method for settlement prediction of single pile and pile group using a hyperbolic model. International Journal of Civil Engineering Vol. 12, No. 2, Transaction B: Geotechnical Engineering, pp. 146-159
- Vesic, A.B. (1963) Beams on Elastic Subgrade and the Winkler's Hypothesis. Proceedings of the 5th International Conference of Soil Mechanics, pp. 845-850

## 3.5. Horizontal bedding moduli of foundation slab

In the properties dialog of the *Foundation slab* a new option has been added.

From now, besides the vertical bedding modulus, the *horizontal* one can be specified by the User, instead of making them equal to the vertical one. These three values define the motion spring stiffness of the automatically created surface support group under the foundation slab. By default, the horizontal values are the half of the vertical one.



This modification has a strong connection with the recently added Pile element. If User would like to model a pile group connected to a foundation slab, the reduction of the horizontal stiffness of the foundation slab can be reasonable. As piles have horizontal supporting directly by the surrounding soil, they are able to carry either the total or a portion of the total horizontal loading.

## 3.6. Camber simulation option for Beams and Profiled plates

A new option has been implemented for beams and profiled plates thatmakes easy to simulate camber of the elements. It can be found in *Beam* and *Profiled plate dialog/General/Camber simulation by prestressing.* 

Beam/Column	Profiled plate/wall
A.1 General Section	A.1 General CC Section T Material Border
Identifier (.position number) B	Identifier (,position number)
Rotation	Model Eccentricity (display only)
Alpha [9]	Physical model              •             •
✓ Camber simulation by prestressing       Total force [kN]	Transverse flexural 1.00
ey'[m]	Camber simulation by prestressing
	e [m]
OK Cancel	Display stiffness > OK Cancel

To calculate camber a camber-type load case needs to be defined. The effect is calculated as a kinematic load.

ol	Name	Туре	Duration class (EN 1995 1-1)		ОК	
1 Ca	mber	+Camber sim. 🔻	Permanent		Cancel	
		Ordinary +Struc. dead load +Soil dead load +Sbrinkage			Import / Export >	
		+Camber sim.	Eurocode code: Load	cases -	Camber - (U) - Translational	displacements - Graph - [mm]
		+Fire +Seis load, Fx+Mx +Seis load, Fx-Mx +Seis load, Fy+My +Seis load, Fy-My				
				$\nearrow$		
				/		



Camber calculation gives statically correct result only for non-eccentric, hinged beams and shells. This only corrects the displacement result and no internal forces will rise in the model.

# 3.7. Easier definition of column corbel's load position

The column corbel's load position is measured from the column axis, instead of the edge of the corbel, which is easier to define.

In version 16:

Corbel	×
A.1 General I Section E End of Identifier (.position number)	
Geometry	
90°       2'       0°         180°       0°       10°         270°       10°       10°         "d" and "e" are measured from physical axis of column (center of gravity of column cross-section)	Alpha [°]       0.000 ∨         ✓ Automatically adjust "d" to bounding rectangle of column         d [m]       0.200         I [m]       0.300         e [m]       0.000         x [m]       0.150         y [m]       0.000
	OK Cancel

In version 17:

Corbel	X
A.1 General Section	laterial End conditions
Identifier (.position number)	
Geometry	
90° 180°- 180°- 90° 180°- 180°- 90°	Alpha [°] 0.000 V Automatically adjust "d" to bounding rectangle of column
	d [m] 0.200
	e [m]
	x [m]
	y [m]0.000

## 3.8. Post-tensioned cable

#### 3.8.1. General

Modelling

Post-tensioned cable object (hereinafter PTC) is a structural component, modelled by equivalent load system.

 $\wedge$ 

Currently the unbonded structural design modelling is available.

The object contains of the shape (continuous line), the reference line (dashed line) and an arrow marking the active end.



The following figure shows a post-tensioned beam after jacking: the actions on the cable a), the actions on the structure b) and the modelled forces acting on the reference line c).

Besides the equivalent forces in the local z direction from the a) deviation, angular the neglected x' component of the angular deviation, the friction force, and moment caused by b) the friction force - cable eccentricity (relative to the could reference line) be significant along the reference c) line.



The plane of the shape can be modified by using "Change direction" or "Rotation" functions.

The definition of a PTC automatically creates two load cases:

- PTC T0: Initial stress state (after the jacking process, it contains the short term stress losses)
- PTC T8: At the end of the design lifetime (it contains the time-dependent stress losses)

#### Definition process

Post-tensioned cable command can be launched from Structure/Bar component/Post-tensioned cable and Structure/Shell component /Post-tensioned cable.



After clicking *Post-tensioned cable* the *Tool window* shows up and User needs to pick the parent object (bar or shell), then define the line of the PTC. The two *Tool window* work slightly different: after the shell selection the user can define multiple cable.



It is strongly recommended to use *Axes*: all aid functions (*Macros*, *Shape wizard*, *Layout wizard*) highly rely on them. It could greatly speed up the definition process.

#### Properties dialog: General tab

By clicking Default settings or Properties the Post-tensioned cable dialog opens.

Post-tensioned cable		X	
A.1 General Shape Result	Manufacturing		
Identifier (.position number)	PTC		
Strands	Short term losses (T0)		
Type	Curvature coef. [-]         0.05           Wobble coef. [1/m]         0.007		
Jacking Jacking side Start then End 🔻	Anchorage set slip [mm]		
Jacking stress [N/mm2] 1488.0	Long term losses (T8) Creep stress loss [N/mm2]		A.1 General Shape Result
	Relaxation stress loss [N/mm2] 0.0 Estimate T8		Strands
Start = (A) (	B) (C)	(D) = End	Type
40 160 -160 0.00 2.50 5.00 7.50	10.00 12.50 15.00 17.50 20.00 22.5 Length [m]	50 25.00	Jacking side
Add to documentation	OK	Cancel	Start then End End then Start

Strand type can be selected from the strand library. The strand library can be edited by the User by clicking *Strands/Type/Edit library*...

Strand library		×
Y1860S7-15,7-F1-C1 Y1860S7-15,7-F1-C2	Name	7-F1-C2
Y1860S7-15,7-F1-C3	f pk [N/mm2]	1860.0
	A p [mm2]	150
	E p [N/mm2]	195000
	Rho [t/m3]	7.850
	Relaxation class	Class 2 🛛 🗸
	Rho 1000 [%]	2.50
New Modify	Delete Import Export	Close

In the same dialog User can select the Jacking side in the drop-down menu next to *Jacking side*. Jacking side options: Start, End, Start then End, End then Start. The last two are *Both-sided* jacking with same stress, but it was necessary to distinct, because the effect of draw-in could results different stress functions using shorter cables.

Jacking stress is calculated as  $0.8 * f_{pk}$  by default after the Strand type selection.

The cable force is reduced by several losses. Their settings are on the right side of the dialog:

Short term losses (initial, T0):

- friction: It is estimated by EN 1992-1-1 5.10.5.2 (1) (formula 5.45) using the Wobble (k) and Curvature coefficients ( $\mu$ ):

$$\Delta P_{\mu}(x) = P_{max} + (1 - e^{-\mu(\theta + kx)})$$

- anchorage set slip
- elastic shortening

Long term, time dependent losses (final, T8):

- shrinkage of structure
- creep of structure
- relaxation of post-tensioned cable

The *Elastic shortening loss* and *Long term losses* can be estimated by the specific dialog.

	Estimate long term losses (T8)
	$\Delta \sigma_{\text{p,c+s+r}} = \frac{\varepsilon_{\text{cs}} E_{\text{p}} + 0.8\Delta \sigma_{\text{pr}} + \frac{E_{\text{p}}}{E_{\text{cm}}} \phi(t, t_0) \sigma_{\text{c,QP}}}{1 + \frac{E_{\text{p}}}{E_{\text{cm}}} \frac{A_{\text{p}}}{A_{\text{c}}} \left(1 + \frac{A_{\text{c}}}{I_{\text{c}}} z_{\text{cp}}^2\right) \left[1 + 0.8\phi(t, t_0)\right]}$
	Data to apply EC2 5.10.6. (2) (formula 5.46)
Estimate elastic shortening loss	Delta sigma pr [N/mm2] 102.06
Data to apply EC2 5.10.5.1. (2) (formula 5.44)	sigma c,QP [N/mm2]
Number of strand on the structure (Plate: on 1m) [-]	Shrinkage [‰]
	Creep c. [-] 0.00
Avarage stress in the strand [N/mm2] 1384	E cm [N/mm2] 33000
E cm,t [N/mm2]	A c [mm2]
A c [mm2]	I c [mm4] 1518750000
I c [mm4] 1518750000	z cp [mm]
z cp [mm]	
Calculated stress values	Calculated long term losses (T8)
Avarage stress in the structure [N/mm2] 27.67	Creep stress loss [N/mm2]
Elastic shortening stress loss	Shrinkage stress loss [N/mm2] 0.00
according to different jacking time [N/mm2] 54.50	Relaxation stress loss [N/mm2]
OK Cancel	OK Cancel

Elastic shortening loss can be estimated by *Estimate ES*... button. This dialog is automatically filled with the parent object's data. Calculate stress values are the result of the estimation and the *Elastic shortening stress loss* field is applied on the *General tab* if the User accepts. It uses modified EN 1992-1-1 5.10.5.1. (2) (formula 5.44) to handle sparsely placed cables:

$$\Delta \sigma_{el} = E_p \sum \left[ \frac{j \Delta \sigma_c}{E_{cm}} \right]$$
  
if  $n \ge 2$  than  $j = \frac{n-1}{2n}$   
else  $j = 0.5n$ 

Where n is the Number of strand.

Average stress in the structure ( $\sigma_c$ ) is informative only.

Long term losses can be estimated by the *Estimate T8*... This dialog works as the previous one: fields are filled with the parent object's data and the calculated results are at the bottom (Relaxation information is in the Strand library). It uses EN 1992-1-1 5.10.6 (2) (formula 5.46). The summation of the calculated long term losses is equal to the result of this interaction formula.

Estimation dialogs calculate cross section data for 1m wide stripes if plate object is the parent.

Both estimation dialog is available during the *Default settings* editing, edit fields are filled with zeros.

Properties dialog: Shape tab

On the Shape tab there are settings related to the geometry of the cable.



The Shape table can contain *Base points* and *Inflection places*: these determine whether linear or parabolic shape is applied.

*Base point*: A point with exact position, user known x' - z' coordinates and the angle of the tangent. Usually minimum and maximum places and end points. (Black rectangles in the preview.)

*Inflection place*: A function connection place  $(x_{inf})$ , where user defines x coordinate only. Using it between Base points determines two parabolic function:  $f_n$  and  $f_{n+1}$  where  $C^1$  continuity is fullfilled:  $f_n(x_{inf}) = f_{n+1}(x_{inf})$ ,  $f'_n(x_{inf}) = f'_{n+1}(x_{inf})$ . (Blue circles in the preview.)



"*Inflection place*" is not an accurate mathematical expression in this feature: the change of function convexity is not ensured. However this term makes recognizable the underlying function.

For x' and z' fields some macros are available:

- x' macro: Start, End, Axis distances
- z' macro: Top, Middle, Bottom

The *Top* and *Bottom* values mean the distance between the axis of the cable and the edge of the element. These values are used by the *z'* macro and Shape wizard.

Using *Shape wizard* (*Shape wizard*... button) parabolic shapes can be easily created. This tool highly relies on previously defined axes. The *Axes* define *outer* (marked by *o*) and *inner* spans (marked by *i*) along the PTC. At the span ends *Base point* 

<b>T</b>	Manufacturin	g		
	Tangent	-	Top [mm]	60.0
5	0.0		Bottom [mm]	60.0
0	0.0		Change winned	
0	0.0		Shape wizard	
0	0.0		Sort by x	
0	0.0			

will be created with maximal z' coordinate, the minimal (*B*, *E*) place of the *Base points* and Inflection places (*A*, *C*, *D*) can be defined by ratio of spans (*Lo* and *Li*). *B/Lo* and *E/Li* minimum positions can be calculated automatically by checking the box next to *Auto calc. min. position*. At that time these textboxes are disabled.



The same settings can be found in the *Tool window* as well for easier definition:

Post-tensioned cable for bars			
Parabolic shape by axes	A/Lo (infl.) 0.10	D/Li (infl.) 0.20	
Reverse shape	B/Lo (min.) 0.45	E/Li (min.) 0.50	
V Auto calc. min. position	C/Lo (infl.) 0.20		

It is not recommended to use Beam/Apply default physical alignment option for beam containing PTC, since it can cause unnecessary eccentricity in the shape because Shape wizard uses the physical element.

Beam	×
🛄 ? 🗅 🗊 🕇 th 🛨 🔽 😳 🗘 🖸	
Apply default physical alignment	

By allowing the *Display physical element* option he section of the cross-sectioned element is shown.





Display physical element option shows all cross-sectioned element, the parent object is signed by dark-grey, the non-parents are signed by light-grey.

The Equilibrium status and Minimal radius of curvature are also displayed.



Properties dialog: Result tab

Result tab is implemented to make easier the checking process. There are several results -listed in the order of calculation - which can be chosen from *Result tab/Result drop-down menu*.



Stress functions can be displayed in fpk ratio by Display stress functions in f pk ratio option.





# The example shows a typical Stress function in the T8 state and an Equivalent force at T0 state.



#### Properties dialog: Manufacturing tab

On this tab the Manufacture drawing can be set with proper x' and z' shift.



The Manufacture drawing can be exported into AutoCAD by clicking Export...

File <u>n</u> ame:	
Save as <u>t</u> ype:	AutoCAD drawing (*.dwg)
Alide Folders	Save Cancel

To fine the division section points can be inserted along the cable. Click *Generate points* to open the dialog. At the third option the division-distance can be set by the User.



The position of each point can be edited with *Edit points* button. By clicking *Sort* button the points will be sorted by their x' value in increasing order.



#### Documentation

By clicking Add to documentation... User can decide about the content of the documentation.

Table of content	Appearence
<ul> <li>General data</li> <li>Radius of curvature</li> <li>Angular deviation</li> <li>Stress function with friction losses</li> <li>Stress function with anchorage set slip losses</li> <li>Stress function with short term losses (T0)</li> <li>Stress function with long term losses (T8)</li> <li>Equivalent force for 1 strand (T0)</li> <li>Equivalent force for 1 strand (T8)</li> <li>Equivalent force for cable (T0)</li> <li>Equivalent force for cable (T8)</li> <li>Manufacture drawing</li> </ul>	Graphs width [mm] 148 Graphs height [mm] 65 Text AaBbYyZz → Graph AaBbYyZz → Display stress functions in f pk ratio



The example shows the first page of a typical PTC documentation.

#### P.1.PTC.1 Post-tensioned cable details

General data Strand type: Y1860S7-15,7-F1-C1 Strand No.: 3 Jacking stress: 1488.0 N/mm<sup>2</sup>; 0.800 \* f pk Jacking side: Start Curvature c.: 0.050 Wobble c .: 0.007 1/m Anchorage set slip: 6.0 mm Elastic shortening loss: 0.0 N/mm<sup>2</sup>; 0.000 \* f pk Creep stress loss: 0.0 N/mm<sup>2</sup>; 0.000 \* f pk Shrinkage stress loss: 0.0 N/mm<sup>2</sup>; 0.000 \* f pk Relaxation stress loss: 0.0 N/mm<sup>2</sup>; 0.000 \* f pk Length (projected): 8.000 m Length (real): 8.026 m Stress T0 at Begin: 1319.5 N/mm<sup>2</sup>; 0.709 \* f pk Stress T0 at End: 1341.7 N/mm<sup>2</sup>; 0.721 \* f pk Stress T0 avg: 1330.6 N/mm<sup>2</sup>; 0.715 \* f pk Stress T8 avg: 1330.6 N/mm²; 0.715 \* f pk

#### Angular deviation function


There are three connected lists available:

- post-tensioned cables: Major and General tab data
- post-tensioned cables secondary data: Stress status and quantity estimation
- post-tensioned cables manufacture table: Manufacture drawing coordinates

List tables		×
Structure	^ Structure	List objects All ~
	Data Concrete materials Sections Beams Plates	AaBbYyZz ▶
Moving load maximum     Imperfections     Load combinations     Max. of combinations     Maximum of load groups	Crucido Post-tensioned cables Post-tensioned cables secondary data Post-tensioned cable manufacture table	Hide irrelevant tables Regional decimal symbol instead of

Post-tensioned cables

ID	Strand type	Strand No.	Jacking stress	Jacking side	Curvature c.	Wobble c.
[-]	[-]	[-]	[N/mm <sup>2</sup> ]	[-]	[-]	[1/m]
P.1.PTC.1	Y1860S7-15,7-F1-C1	3	1488.0	Start	0.05	0.007

Anchorage set slip	horage set slip Elastic shortening loss Creep stress loss		Shrinkage stress loss	Relaxation stress loss	
[mm]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[N/mm²]	[N/mm²]	
6.0	0.0	0.0	0.0	0.0	

Post-tensioned cables secondary data

ID	Strand type	Strand No.	fpk	Stress T0 S	Stress T0 E	Stress T0 Avg
[-]	[-]	[-]	[N/mm²]	[N/mm²]	[N/mm²]	[N/mm <sup>2</sup> ]
P.1.PTC.1	Y1860S7-15,7-F1-C1	3	1860.0	1319.5	1341.7	1330.6

Stress T8 Avg	Min. radius of Curvature	Length (projected)	Length (real)	Length (all strand)	Volume
[N/mm <sup>2</sup> ]	[m]	[m]	[m]	[m]	[m3]
1330.6	27.719	8.000	8.026	24.077	0.004

Mass
[t]
0.028

Post-tensioned cable manufacture table

ID	x' shift	z' shift	Point ID	X'	z'
[-]	[mm]	[mm]	[-]	[m]	[mm]
P.1.PTC.1	0	100	1	0.000	240
			2	1.000	122
			3	2.000	32
			4	3.000	-22
			5	4.000	-40
			6	5.000	-22
			7	6.000	32
			8	7.000	122
			9	8.000	240

Handling multiple PTCs properties

If the PTC-s have the same settings, length and shape, all features of properties dialog are available, otherwise only the common properties appear.

Post-tensioned cable	Post-tensioned cable
A.1 General         Shape         Immit Manufacturing           Identifier (postion number)         Identifier (postion number)         Identifier (postion number)	A.1. General         Shape         Emeral         Manufacturing           Result         Equivalent force for PTC (10)
Strands         Short term losses (10)           Type	Preview is not shown: Multiple cables were selected with different geometry.
Preview is not shown: Multiple cables were selected with different geometry.           Add to documentation         OK         Cancel	Preview is not shown: Multiple cables were selected with different geometry.  Add to documentation  OK Cancel



*Add to documentation...* and *Manufacture tab/Export...* functions are available at multiple selection and they document all selected cables at the same time.

Others

Post-tensioned cable option is added to the Colour schema. Available colour modes: ID, Strand type, Strand number, Jacking side.

PTCs can be selected by several option of the Filter: Structural element, Identifier, Strands

PTCs have detailed tooltips supplemented by length and Stress status (connected chapters from the EN 1992-1-1: 5.10.2.1 (1) and 5.10.3 (2)).

Post-tensioned cable: P.1.PTC.0
Strand type         Y1860S7-15,7-F1-C1           Strand number         1           Jacking side         Start
Jacking stress [N/mm2] 1488.0 (0.800 * f pk)
Short term losses           Curvature coef. [-]           Wobble coef. [1/m]           Anchorage set slip [mm]           Elastic shortening loss [N/mm2]
Long term losses Creep stress loss [N/mm2] 10.6 Shrinkage stress loss [N/mm2] 3.9 Relaxation stress loss [N/mm2] . 85.1
Length Projected [m]
Stress T0 at Start [N/mm2] 1365.3 (0.734 * f pk)
T0 at End [N/mm2] 1404.1 (0.755 * f pk) T0 avg [N/mm2] 1406.8 (0.756 * f pk) T8 avg [N/mm2] 1307.3 (0.703 * f pk)

#### 3.8.2. Layout wizard

It is a parametric tool to create a set of PTCs in a specific layout at the same time. It can be launched from *Structure/Shell component/Post-tensioned cable/Layout wizard*.

Post-tensioned cable for shells		
J ?   🖊 🗖 🧮		
Parabolic shape by axes A/Lo (infl.) 0.10 D/Li (infl.) 0.20		
Reverse shape B/Lo (min.) 0.45 E/Li (min.) 0.50		
Auto cilc. min. position C/Lo (infl.) 0.20		
	9 4 J 4 J 4 J	
	Post-tensioned cable layout wizard	×
	Load settings	Uneaqual loading
	Considered load [kN] Struc. dead load! 2600.0	Column span load ratio [%]
Select plate	Balance ratio of the load [%]	Column span width ratio [%]
plate	Maximal deviation [%] 10	Different strand number
-		
	Direction and shape settings	Placement
	Direction and shape settings Used direction $\bigtriangledown \mathbf{x}' \qquad \bigtriangledown \mathbf{y}'$	Placement PTC - PTC minimal distance [mm] 300
	Used direction	PTC - PTC minimal distance [mm] 300
	Used direction	PTC - PTC minimal distance [mm] 300 PTC - PTC maximal distance [mm] 1500
	Used direction         ✓ x'         ✓ y'           Minimal strand number         1         1           Maximal strand number         3         3	PTC - PTC minimal distance [mm] 300 PTC - PTC maximal distance [mm] 1500 PTC - Column minimal distance [mm] . 300
	Used direction         I         Y           Minimal strand number         1         1           Maximal strand number         3         3           Top [mm]         60         60	PTC - PTC minimal distance [mm]         300           PTC - PTC maximal distance [mm]         1500           PTC - Column minimal distance [mm]         300           Minimal length [m]         1.00
	Used direction         y'           Minimal strand number         1           Maximal strand number         3           Top [mm]         60           Bottom [mm]         60           Shape wizard using height [mm] instead of         200	PTC - PTC minimal distance [mm]         300           PTC - PTC maximal distance [mm]         1500           PTC - Column minimal distance [mm] .         300           Minimal length [m]         1.00

The *Layout wizard* searches the closest solution to the *Unequal loading* settings from the PTC layout variations which fulfilled the following conditions:

- difference between the sum of the PTC's forces acting in local z+ direction and the product of *Considered load* and *Balance ratio* is less than the *Maximal deviation*
- geometrical requirements

The Layout calculation process shows the currently calculated layout parameters and a few indicator values. After the calculation it shows the parameters of the best found solution.

Layout calculation	×
Initial data approach	
Load [kN]:	7620.6 / 6750
Balance ratio [%]:	112.9 / 90
Deviation [%]:	22.9 / 10
Column span load ratio [%]:	50.1/55
Layout status	
Cable number:	104
Estimated avg. sigma x' [N/mm2]:	1.46
Estimated avg. sigma y' [N/mm2]:	1.57
Equivalent transversal loads x'/y' ratio:	1.48
Progress: 148 / 165	
Apply	Cancel

The function uses *Axes* to divide the selected structural element to stripes (Mid span and Column span) assuming columns in the axes cross points.



The algorythm can handle the holes of the plate regions.



At least two Axes crossing the plate are needed to use this function.



Checking Shape wizard using height [mm] instead of physical model option could highly reduce the runtime if the selected plate's thickness is constant.

## 3.9. "No shear" edge macro for Plane walls

"*No shear*" macro has been added to the *Edge* list of *Plane wall* edit tools. Using it defines edge connection with predefined "*No shear*" rigidity type on the defined wall's bottom edge.

ſ	Plane wall				
		• 🗆 🗇 🖬 🔽	'eee 🗆		
		h [m] 3.0000	t1 [m] 0.20000 🕽	Alignment	$\longrightarrow \bullet$
		Placement 🔘 Above		Edges	
		elow		Alpha [º]	Rigid
	1		* / *	1~~~	No shear

It is the recommended wall definition method to create a non-bracing system element.



The following example shows a displacement graph of a multi-storey building where W.4 wall is not part of the bracing system.



Line connection library		×
No shear	Name	. No shear
	Motions [kN/m/m]	Plastic limit forces [kN/m]
	Compression Tension	Compression Tension
	Kx' 0.000e+00 ▼ → 0.000e+00 ▼	□ 1.000e+15 □ 1.000e+15
	Ky'. 1.000e+07 ▼ → 1.000e+07 ▼	1.000e+15
	Kz' 1.000e+07 ▼ → 1.000e+07 ▼	1.000e+15
	Rotations [kNm/m/º]	Plastic limit moments [kNm/m]
	Compression Tension	Compression Tension
	Cx' 1.745e+05 ▼ → 1.745e+05 ▼	1.000e+15 1.000e+15
	Cy' 1.745e+05 ▼ → 1.745e+05 ▼	1.000e+15
	Cz' 1.745e+05 ▼ → 1.745e+05 ▼	1.000e+15
	Predefined types	Behaviour
		Detach No 🔻
		Friction factor
New Modify Delete	Import Export	Close

"No shear" rigidity type settings with free local x' motion rigidity:

## 3.10. Wall corbel

Wall corbel is a new structural element. It can be found in the Structure tab/Shell component/Wall corbel.

	<b>₫</b>	
Shell		
	Wall cor	bel

In the tool window it can be set whether the corbel should be adjusted to the positive or negative side of the wall according to its local coordinate-system.

Wall corbel		×
P	Side Plus  Minus Plus	

Wall corbel	×	Wall corbel
A.1 General Connection Mat	terial	A.1 General Connection
Type		Identifier (.position number)
Motion springs [kN/m/m]	Plastic limit forces [kN/m]	
Compression Tension	Compression Tension	Geometry
Kx 1.000e+07 ▼ → 1.000e+07 ▼	1.000e+15 1.000e+15	Side Plus 🔻
Ky' 1.000e+07 ▼ → 1.000e+07 ▼	1.000e+15	
Kz' 1.000e+07 ▼ → 1.000e+07 ▼	1.000e+15	
		+ + - n1 [m]
Rotation springs [kNm/m/°]	Plastic limit moments [kNm/m]	h2 [m] 0.600
Compression Tension	Compression Tension	x [m] 1.000
Cx' 1.745e+05 ▼ → 1.745e+05 ▼	1.000e+15     1.000e+15	
Cy' 1.745e+05 ▼ → 1.745e+05 ▼	1.000e+15 1.000e+15	
Cz' 1.745e+05 ▼ → 1.745e+05 ▼	1.000e+15	
Predefined types	Behaviour	
	Detach No 💌	
	Friction factor 0.300	
Local system		
Keeps direction along line	z' y'	
	OK Cancel	OK Cancel



À

Wall corbel can be placed on Plane wall only.

#### 3.11. Editable surface connection (soil)

Stiffness properties of automatically generated surface connections are editable.



The example shows a settled foundation slab into soil.

Right click on the surface connection, then select *Properties* (see chapter 2.1.).



## 4. Load

## 4.1. Construction stages

The Construction stages function helps to model and design the different building phases, and it helps to determine the construction process affected displacement/internal force distribution of the finished structure. **Currently we implemented this feature for high-rise buildings, making the structural element separation by the storey system, but it will be generalized to a user-defined stage system in the near future.** 

Storeys must be defined to use Construction stages, since currently one storey is considered as one Stage.

Construction stages can be defined under the load tab. Click "Generate" button to create stages with the correspondent storey name.



Columns in the table mean the following:

- No: Number of the stage
- Stage description: Name of storey which is built in the stage
- *I* (*Initial stress*): if checked, in calculation of the stage displacements will be reset to zero, but other results (internal forces) are accumulated
- Activated load cases: Activated load cases for the stage
- Partitioning:

This defines for each load from the load cases that how and when it is activated during the construction stage calculation.

• *only in this stage*: Load case is activated in the specified construction stage and acts only in this stage.



Remaining loads from these load cases will not activate in other stages.

#### Only in this stage (applied in 2nd stage)



 from this stage on: Load case is activated in the specified construction stage and acts in this and also in the remaining stages – the parts of the loads that act on the previous storeys are also activated in the specified stage

#### From this stage on (applied in 2nd stage)



 shifted from first stage: Load case is activated in the specified construction stage and acts in this and also in the remaining stages – the parts of the loads that act on the first storey will act in this specified stage and the remaining parts of the loads that act on the second storey will act in the next stage and so on (e.g.: flooring and covers)

## Shifted from first stage (applied in 2nd stage)



It is possible to add any construction stage to any load combination with the following limitations:

- Only one construction stage is allowed in one combination.
- Cannot combine a construction stage and a load case which is already activated in a construction stage
- The fire and/or seismic load cases can be combined with only the final construction stage

For load groups only the final construction stage can be added too.



The construction stages automatically follow all storey modifications.

User can start the construction stage calculation at Analysis/Calculation/Construction stages. There are two calculation methods, so called *Incremental "Tracking" method* and *"Ghost" structure method*.



When incremental method is chosen, the model is built stage-by-stage. In case of "ghost" structure method the full structure is in the calculation, but stiffness of those structural parts which aren't in the specific stage is highly reduced.









The construction stage results can be found in the New results/Analysis/Construction stages. For every stage result the method name and the displayed construction stage (e.g. CS.1 Storey 1) appears in the information panel.

Adding any new Construction stage result will open a "Construction stages" (result-display) tool to make easier the navigation between the stage results and to animate the Construction process if it is needed.



It's also possible to choose the construction stage in detailed results.

FEM-Design 17 - 3D Structure - Untitled File Edit Draw Modify Tools Settings View	w Window Help		
Detailed result			
		1.00 + 1.1 1.0 Outok change	
	Action (1.70) 1.70 1.72 0.85 0.43 -0.43 -0.43 -0.43 -1.70	↑fi2' [0]           0.0127           0.0043           -0.0043           -0.0043           -0.0043           -0.0172	

The equilibrium dialog contains the construction stages, too.

Equilibrium				×
Check	stages 🗸 IJ	- Ultimate	~	Close
Load cases / cor			jes	
CS.1: Storey 1 CS.2: Storey 2 CS.3: Storey 3 CS.4: Storey 4				
Component	Loads	Reactions	Error [%]	
Fx [kN]	0.00000	0.00000	-	
Fy [kN]	0.00000	0.00000	-	
Fz [kN]	-120.00	80.000	33.33	
Mx [kNm]	0.00000	0.00000	-	
My [kNm]	1840.0	-480.00	73.91	
Mz [kNm] [	0.00000	0.00000	-	

It's possible to list the construction stages result, which can be found under Analysis/Construction stages.



## 4.2. Pre defined Psi values for temporary load groups

There is an option for temporary load groups to choose predefined  $\psi_0$ ,  $\psi_1$  and  $\psi_2$  values.

	Load grou	ıps		~ .		/ =_ /		_	×
N	0		Load	l group			Included load cases	^	ОК
	_								Cancel
	Nam Type	e Tem	oorary		~				Import / Export >
	Safe	ty factor		1.50					Combination method
	Psi 0 Psi 1			0.70	Π				● EC0 6.10 ○ EC0 6.10.a, b
	Psi 2			0.20	2				Load group
Description		Psi 0	Psi 1	Psi 2	~				Insert
Imposed loads		0.70	0.50	0.30					Delete
Category A: domestic, residential areas									Delete all
Imposed loads Category B: office areas		0.70	0.50	0.30					Load case
									Insert
Imposed loads Category C: congregation areas		0.70	0.70	0.60					New
								~	Remove
Imposed loads Category D: shopping areas		0.70	0.70	0.60					
					~				
		OK		Cancel					

## 4.3. Ignored temporary load cases

Temporary load groups has an option to ignore in SLS combinations in Load group maximum results and in Generating load combinations by load groups.

Name
Type Temporary V
Safety factor 1.50
Psi 0 0.70
Psi 1 0.50 >
Psi 2 0.20
Potentially leading load cases
☑ Ignore in SLS combinations
OK Cancel

#### 4.4. Deviation load improved

There are changes in deviation load macro.

- User can set the value of "m" instead of "alpha m", which is calculated automatically from "m". The applied values can be given for each storey, which will be stored.
- A checkbox has been implemented, which allows generation of surface load instead of point load.

- Loads can be defined in positive and/or negative direction, which can be applied with the *plus/minus buttons* next to the icons , in the bottom-right corner of the dialog.



As many load cases will be generated as many buttons are checked.

Storey	Level [m]	^ No	Load case	Factor	^	Considered force on storey [kN] 0.000
orey 1	3.00		1 Point load	0.00		alpha h (proposed value)
orey 2	6.00		2 Line load	0.00		
orey 3	9.00		3 Surface load	0.00		alpha h (applied value) 2/3 <= x <= 1.0
			4 Line temperature variation load	0.00		
			5 Surface temperature variation lo	0.00		m (proposed value) 2
			6 Line stress load	0		m (applied value) 2
			7 Surface stress load	0	•	m (applied value)2
			8 Point support motion load	0.00		alpha m 0.866
			9 Line support motion load	0.00		Lond units Ref. 0.000
			0 Surface support motion load	0.00		Load value [kN]
		:	1 Mass	0.00		
			2 Wind load X+	0	3.	① ,   ∓   ∓   1 ,   ∓ ∓
	2.	v -	3 Wind load Y+	0	, V	

## 4.5. Notional load

This function generates horizontal load from non-horizontal loads using a conversion factor.

It can be reached in Load tab/Macro/Notional load.

	Ľ	ን 🛆 🔶	⚠ 🗖	ł			
		Macro	Notional				
Notional load	1.		×				×
Notional load case name	Notion	al		No	otior	nal	
Conversion factor [%]0.15	No	Load case	Factor	1	٧o	Load case	Factor
Considered Z-direction loads only		case 1	0.00			. case 1	1.00
		case 2	0.00			case 2	0.00
		case 3 case 4	0.00			case 3 case 4	0.20
	-	case 4				Notional X+	0.00
2.	<u> </u>		=		6	Notional X-	0.00
					7	Notional Y+	0.00
EC seismic notional load				TL	8	Notional Y-	0.00
According to EC0 A1.3.2							
Generate 3.			Close				Close

- For the notional load case a name can be given by typing it into the textbox on top of the dialog
- After clicking *EC seismic notional load* button, Load case factor will be set according to parameters of Load groups the Load case belongs to. It uses EN 1990-1-1 A1.3.2 DK NA:  $A_d = 1.5\% (\sum G_{k,j}" + "\sum_{i\geq 1} \psi_{2,i} Q_{j,i})$
- After clicking on the *Generate button*, as many *Notional Load cases* will be created, as many directions are chosen above the *Generate button*

The generated *Notional Load Cases* appear in the *Load Cases dialog* and they can be selected from *Loads tab/Current load case* drop-down list.

	1	1	1				
No	Name	Type	Duration class (EN 1995 1-1)	Â	ОК		
1	case 1	Ordinary	Permanent		Cancel	10	ads Finite elements
2	case 2	Ordinary	Permanent				
3	case 3	Ordinary	Permanent		Import / Export >	14	Notional Y-
4	case 4	Ordinary	Permanent				
5	Notional X+	Notional	Permanent		Insert		case 1 case 2
6	Notional X-	Notional	Permanent	=			case 3
7	Notional Y+	Notional	Permanent		Delete		Notional X+
8	Notional Y-	Notional	Permanent		Delete all		Notional X-

The process of the load generation by Notional load macro is shown on the picture below.



#### 4.6. Load comments

A new property – *Comment* - has been added to all load types. With this new feature the User can label every single load in order to easily identify them.

The comments can be set for every load types in the Default settings/Properties dialog.



Load comments can be turned on/off by Settings/Display/Load dialog's Display comment option.

Settings	Load	
Calculation     Display     Display	Graphical options	Temperature variation           Size/Scale [m]         1.0000           Stress load         1.0000           n-Size/Scale [m]         1.0000           m-Size/Scale [m]         1.0000           Support motion         1.0000           Size/Scale [m]         1.0000           Moving load         1.0000           Local s. size [m]         1.000
Save as default		OK Can

Load comments appear in the Filter dialog and also in the listed load tables.

Filter	x						
- Load comment - force-x - force-y - force-z - force-Térbeli - moment-x - moment-y - moment-z - moment-Térbeli - Force-x - Force-y - Force-z - FORCE-x	* III						
MSSZ	-						
Reset							
OK Continue Cancel							

Point loads

No.	F	М	Load case	Comment
[-]	[-] [kN] [kl		[-]	[-]
1	10.000	0.000	force	force-x
2	10.000	0.000	force	force-y
3	10.000	0.000	force	force-z

## 4.7. Load export / import via clipboard

Load Export and Import via clipboard is implemented, to let the user easily and quickly modify loads.



In order to export loads, User has to click to *Export* to send the load information to the clipboard. Then User can paste to Excel or any editor program and modify them.

	Ŷ	
L	ł	7

Only comments and the load intensities can be modified. We suggest NOT to edit other columns to avoid errors in Importing.

			$\checkmark$	V	,	+			1		
#Type	Guid	Case	Comment	q1 k	N/m2	q2 [kN/m2]	q3 [kN/m2]				
LDSURFFORCE	580001db-8b57-4a8e-945f-fc957f1b8d8f	force	FORCE-x		10	20	) 3	0			
LDSURFFORCE	bb8be40c-cc6c-4cc9-aab0-8727b3c74141	force	FORCE-x		10	10	) 1	0			
LDSURFFORCE	9c620550-5321-402c-8846-8434b40ef5c6	force	FORCE-x	Ý	10	20	) 🗸 3	0			
#Type	Guid	Case	Comment	Z0 [ m	ן ו	q0 [kN/m2]	qh [kN/m2/m	]			
LDSURFFORCESP	33da63d3-965b-4035-9130-3957241e1237	force			0	(	)	1			
LDSURFFORCESP	40c74961-da03-43e4-a590-64efc8538c61	force		- V	0	(	)	1	Y		
#Type	Guid	Case	Comment	n1 (k	N/m]	n2 [kN/m]	n3 [kN/m]	m1 [k	Im/m]	m2 [kNm/m]	m3 [kNm/m]
LDSURFSTRESS	3a03da68-48a4-4478-9fa2-9b6ac4217950	stress	const	1	10	10	) 1	0	10	) 10	10
#Type	Guid	Case	Comment	e1 (n	n]	e2 [m]	e3 [m]				
LDSURFSUPP	ad0e8d96-c1a1-43ab-97ac-030d2cc9eee1	supp motion	Ey		0.01	0.01	. 0.0	1			
LDSURFSUPP	063863f6-1dd1-46e3-8bb4-ebd491de6cb0	supp motion	Ex		0.01	0.01	. 0.0	1			
LDSURFSUPP	47870d6b-9c4a-4767-9a03-d72784424359	supp motion	Ez	↓	0.01	0.01	. 0.0	1	↓ _		
#Type	Guid	Case	Comment	t1 [°C	]	t2 [°C]	t3 [°C]	t1'[°0	]	t2' [°C]	t3' [°C]
LDSURFTEMP	bc68facf-a07a-4ed5-993f-bbd5c4c58736	temperature	CONST_circle		10	10	) 1	0	1(	10	10
LDSURFTEMP	b9b06ce8-8bda-48b9-a492-4bc790ca3d8e	temperature	CONST		10	20	) з	0	1(	10	10
LDSURFTEMP	f4f7f053-a12a-46f6-9798-8ef647ae53e3	temperature	CONST		10	10	) 1	0	10	10	10

After changing attributes, User can choose whether to import some. or all of the loads by selecting the desired rows and copying them to clipboard, then in FEM-Design clicking on *Import*.



If the User exported constant surface load, only changing the first intensity value will have effect on the surface load.

# 5. Analysis

#### 5.1. Plastic trusses, supports and connections

Plastic calculation has been implemented for trusses, supports and connections.



It is also available for edge connections of all shell elements (Plane plate and wall, Profiled plate and wall, Timber plate and wall, Fictitious shell)

Plane plate	8

The plastic data can be set in the *Default settings/Data tab* of each above mentioned options. The figures show where the feature is placed in the three types of dialogs of these items.

Some examples with detailed calculations can be found in the verification book.

Truss	×
A.1 General Section Materia	d
Identifier (.position number)	
Compression	Tension
✓ Limited capacity [kN] 1.000e+15	✓ Limited capacity [kN] 1.000e+15
Behavior	Behavior

The options above are considered only for load combinations calculated as non-linear elastic. *Plastic* behaviour is considered for load combinations calculated as non-linear elastic + plastic. See more details in the next chapter.

For further information check the <u>documentation</u>.

Point motion support	Edge connection	
A.1 General 🛔 Data	A.1 General Data	
	Туре	
Spring [kN/m], [kN]	Motion springs [kN/m/m]	Plastic limit forces [kN/m]
	Compression Tension Kx' 1.000e+07 ▼ + 1.000e+07 ▼	Compression Tension
Compression Tension		
K 1.000e+10 ▼ 1 1.000e+10 ▼	Ky' 1.000e+07 ▼ → 1.000e+07 ▼	1.000e+15
Compression Tension	Kz' 1.000e+07 ▼ → 1.000e+07 ▼	1.000e+15 1.000e+15
Plastic limit forces V 1.000e+15 V 1.000e+15	Rotation springs [kNm/m/º]	Plastic limit moments [kNm/m]
	Compression Tension	Compression Tension
	Cx' 1.745e+05 ▼ → 1.745e+05 ▼	□ 1.000e+15 □ 1.000e+15
	Cy' 1.745e+05 ▼ → 1.745e+05 ▼	□ 1.000e+15 □ 1.000e+15
Setup "Rigid" > OK Cancel	Cz' 1.745e+05 • + 1.745e+05 •	🖾 1.000e+15
Surface support group	Predefined types	Behaviour
	min and the second	Detach
A.1 General Data		Friction factor
	Local system	
Type	Direction changes	
Motion springs [kN/m2/m] Plastic limit forces [kN/m2]	along line	X
Compression Tension Compression Tension		z.4 - y
Kx' 1.000e+05 ▼ → 1.000e+05 ▼ □ 1.000e+15 □ 1.000e+15	Setup "Rigid" >	OK Cancel
Ky. 1.000e+05 ▼ ♦ 1.000e+05 ▼ □ 1.000e+15 □ 1.000e+15		- Cancer
Kz' 1.000e+05 ▼ ♦ 1.000e+05 ▼ □ 1.000e+15 □ 1.000e+15		
Behaviour		
Detach No 💌		
Setup "Rigid" > OK Cancel		

# 5.2. Modified behaviour of trusses in non-linear elastic (NLE) and non-linear elastic + plastic (PL) calculations

Limited tension capacity is available for trusses from now on. In the former versions only limited compression capacity was implemented.

Truss	×
A.1 General I Section Materia	al
Identifier (.position number)	
Compression Culmited capacity [kN] 1000 Behavior	Tension United capacity [k1] 2000 Behavior
Compression, Brittle Compression, P	lastic Tension, Brittle Tension, Plastic

In the earlier versions of FEM-Design in case of NLE (so-called uplift) calculation, if maximal compression capacity of a truss was set to zero, and in the first iteration step compression arose in the truss due to the external loads, the calculation method did not allow tension to arise during the further iteration steps, even if it was theoretically possible. Now it has been implemented that compression and tension are changing during the iteration in case of NLE calculation.

However the actual effect of the two behaviour options (brittle or plastic) depends on the setup of the specific load combination.



## 5.3. Check for identical copies of structural objects and loads

A new option is added to the *Settings/All.../Calculation/Analysis* dialog. User can set if the program should check the identical copies of structural elements and/or loads.



For example, let's take the following example for object multiplication in a model:

- point support: 2 identical supports (1 unnecessary copy)
- line support: 3 identical supports (2 unnecessary copies)
- $\rightarrow$  3 objects, 5 unnecessary copies
- beam: 3 identical beams (2 unnecessary copies)

Having this option ON, during pre-processing of calculation, the following question pops up:

FEM-Design - Question							
?	One or more identical copies of 3 structural elements are found. Do you want to delete all identical copies (5 structural elements) before calculation?						
	<ul> <li>Yes, to delete all identical copies and continue calculation.</li> <li>No, to calculate without deleting.</li> <li>Cancel, to check and fix the problems one-by-one, using "Tools/Correct model/Delete identical copies" function.</li> </ul>						
	Yes No Cancel	]					

The program informs about the number of unnecessary copies. User has three options to select:

- clicking "Yes", FEM-Design will delete all unnecessary copies automatically,
- clicking "No" will ignore this check, and the calculation will runs without any modification,
- clicking "Cancel" will abort the calculation and allows User to fix the *unnecessary objects* with the new feature called *Correct model Delete identical copies*, described in Chapter 1.1.

## 5.4. Fully rigid diaphragm

There is a new type of diaphragm calculation: Fully rigid.

Calculations	×	
Calculations Analysis Bigenfrequencies Seismic analysis A	Finite element types         Finite element types         Fine elements with 9/6/3 nodes. Longer calculation time.         Diaphragm calculation         None         IRigid membrane         Fully rigid         Peak smoothing	
Recalculation	OK Cancel	

Fully rigid diaphragm will ensure the rigid movement of the all objects lying in diaphragm's region and move them all together in one plane in any direction (as rigid body). For further information check the <u>documentation</u>.



## 5.5. Selecting all relevant shapes in Modal analysis

This new feature helps to select all the relevant shapes in the Modal analysis. By clicking the "Select all" button the program selects all the shapes with non zero modal mass percentage.



This feature is not selecting the dominant shapes, it has to be selected manually.

Method Options	etup						×						
		-		Se	t Auto	]							
Static, linear shape		[s] 3.930	mx'[%] 0.0	my'[%] 5.2	mz[%] ^	Include Select a							
<i>H</i>	2	2.093	0.0	92.3	0.0	Select							
H		0.945	0.0	0.0	0.0	Clear							
Static, mode shape		0.640	92.7	0.0	0.0	Dominant s	h	$\langle \rangle$					
		0.606	0.0 3.8	0.0	Seismi	c analysis, se	tup	$\backslash$					
H	Summation ru			0.0	Meth	od Options							
Modal analysis	SRSS: 1	$E_{E} =$	$\sqrt{\sum E_j^2}$	2 Bi		Ħ		analy Alfa [°]	\	Set	Auto		
Add to document	ation				lie.	Static, lear shape	No	T[s]	mx'[%]	my'[%]	mz[%] ^	Indude	
							1	3.930 2.093	0.0	5.2 92.3	0.0	Select all	
						$\square$	3	0.945	0.0	0.0	0.0	Select	
					-		4	0.827	0.0	0.0	0.0	Clear	
					m	Static, ode shape	5	0.640	92.7	0.0	0.0	Dominant sh.	
							6	0.606	0.0	0.0	0.0	Select	
							7	0.550	3.8	0.0	0.0 🗸	Clear	
						<u> </u>	Summatio	n rule by d	$\frac{1}{\sqrt{\sum E_j^2}}$	_		1	

# 6. RC design

## 6.1. RC Bar detailed result reinforcement export to DWG/DXF

Now the detailed results for RC bar reinforcement can be exported into \*.dwg or \*.dxf file format. This feature can be reached in detailed result window, File/Export to/Export to AutoCAD...

29 F	EM-Design 17 -	3D Structure - 2_12.str						
File	Edit Draw M	odify Tools Settings	View Window Help					
۳,	New	Ctrl+N						
p_	Open	Ctrl+O	Mariana A					
	Save	Ctrl+S	Maximum					
	Save as	Ctrl+Shift+S	Manage results					
	Export to		Export to Office OpenXML (.docx) document					
	Recovery files		Export to XHTML+MathML page					
	Print	Ctrl+P	Export to Mathcad 15 worksheet (beta)					
	Exit	Ctrl+Q	Export to AutoCAD (.dwg   .dxf)					
	2_12.str		ad combinations					
	9_LR.str							
	Klári\Infopark_v6	_v16_Soil_mod.str						
	Klári\AdlnBooklet	_v17.str						
	9.str							

## 6.2. RC Bar detailed result drawing improvements

The improvements on the Reinforcement drawing are as follow:

- stirrup's dimension lines
- longitudinal bar dimension lines
- cross -section with reinforcement numbers

## Stirrup dimension lines



## Longitudinal bar dimensions



## Cross-sections with reinforcement members



## 6.3. RC Shell buckling

For the consideration of buckling failure of walls and plates, a new checking criterion is available for RC shells, the *shell buckling*. The buckling problem of the shell is transformed to the buckling of equivalent columns made from the shell, on which the second order resistance and utilization is calculated.



Only RC Plane plates and Plane walls with straight reinforcement and uniform thickness are suitable for shell buckling calculation.

The calculation process is based on so-called buckling regions, which can be defined at *RC design/Surface reinforcement/Buckling length*.



Each buckling region on the shell has a corresponding buckling factor (beta) and a direction vector in the plane of the shell. The former will be used to calculate the buckling length of the equivalent column, while the latter one specifies the x' longitudinal axis of this column. By default, FEM-Design generates one buckling region on each RC wall and plate. Default buckling direction is vertical on walls, and parallel with the local x axis on plates. Buckling factor is set to 0.0 on all shells in order to let the User decide whether this calculation is needed or not, since it is quite time consuming.



Shells with zero buckling factor will not be considered for shell buckling calculation, but zero utilization is set for them.

The default buckling regions can be modified by adding new regions to the shell. One shell may have more buckling regions with different beta factor and direction vector, but the shell must be completely covered by these regions.

Buckling length	x
$ \int_{\underline{x}} \ell_{\circ} $ beta 1.00 >	

During the checking process, the program generates equivalent bar(s) from the shell based on its material, thickness and reinforcement. This bar is checked as an RC bar: Its utilization is calculated by determining its second order internal forces and resistance.

The calculation process consists of the following steps:

1. As other shell design calculations, the shell buckling is also calculated in every node of the shell (only where there is a buckling region with non-zero beta value).



2. An equivalent bar is generated for the examined node as follows. The edges of the shell are intersected by the ray determined by the node and the direction vector of the corresponding buckling region. The two intersection points are taken as the start and the end point of the equivalent bars.



Note that this intersection is always made with the edges of the shell and not with the edges of the buckling region corresponding to the node! If a node is on the border of two or more buckling regions, it is calculated with both different beta values and direction vectors, and the higher utilization will be used.



3. The cross section of the equivalent bar is 1 m wide and its height equals to the thickness of the shell. Along the bar, the applied reinforcement of the shell is transformed into the direction of the bar and placed into it.



The checking process is executed section by section along the bar. The distance between these sections is given by *Division length of substitute column* parameter in Calculation parameter dialog (see the lower figure). Internal forces acting at these sections are calculated by

transforming shell internal forces at the section point into the coordinate system of the column. As the buckling direction of shells is perpendicular to its plane, we need the equivalent bar's normal force and moment vector in the plane of the shell for the calculation.



4. Once the first order internal forces are obtained in every section, the second order internal forces are calculated based on the *nominal stiffness* or *nominal curvature* method, according to the configuration settings. The only difference in the checking process of a real bar and this equivalent bar is that now the eccentricity coming from the second order effects are applied only perpendicularly to the plane of the shell. In other words, the out-of-plane normal force has eccentricity only along the z' axis of the shell. This modification is in harmony with the fact that the buckling direction of the shell is perpendicular to the plane.



5. Finally, based on the second order internal forces, the utilization is calculated for every cross section of the equivalent bar (based on the interaction curve), and the highest section utilization is assigned to the node.

Shell buckling calculations are available for *Load combinations*, *Maximum of load combinations* and *Maximum of load groups*. The utilization results can be displayed in the *New result/RC shell/Shell buckling/Utilization* 

Results ☐ Analysis ☐ Load cases ☐ Load combinations	Utilization Calculaton	Calculation	
Class Control Heads     Constructions     Cabel     Cabel	Catulation Lead combinations V Combination Poissum 1	Objects table           UNLinform           0 m.0         10x1           0 m.0         10x1	- 1000 1000 - 1000 1000 0.5/5 78 78
		Det Dév	CHARGE

Some details of the calculation can be obtained by listing *RC design/Load combinations/Shell, buckling* table. Also, wall buckling utilization appears in the *Shell, Utilization* list.



Shell, Buckling, Load comb.: 1

ID	Utilization	х	у	Z	As, top	As, bot	As, mid	N_Ed
[-]	[%]	[m]	[m]	[m]	[mm2/m]	[mm2/m]	[mm2/m]	[kN/m]
W.1.1	28	25.519	33.973	-3.000	523.599	523.599	0.000	-98.771
W.2.1	144	43.200	26.320	-3.000	523.599	523.599	0.000	-166.218

M_Ed	M2_Ed	N_Rd	M_Rd	Bar length	Beta
[kNm/m]	[kNm/m]	[kN/m]	[kNm/m]	[m]	[-]
9.146	14.733	-353.254	52.693	3.000	2.000
31.101	57.126	-115.298	39.626	3.000	2.000

Every plate and wall has one result, containing the coordinates of the dominant section, the corresponding reinforcement, first and second order internal forces together with the capacity and buckling factor.

## 6.4. RC shell - EN 1992-1-1, Annex F

A new checking criterion is added for RC shells according to EN 1992 Annex F.

This feature is meant to inform the user whether the shell reinforcement is optimal or not according to the EN 1992-1-1, Annex F. The reinforcement is optimal, if the cracks for serviceability limit state are acceptable and the required deformation capacity for ultimate limit state is provided. This method only works for in-plane stress state. FEM-Design calculates the stresses in the reinforcement plane. The program uses the F1.(4) formulas for both the applied and required reinforcement.



Compression is positive in the formulas.

The results can be listed in the RC design for Load Combinations, Max. of combinations and Maximum of load groups.



If the Danish national annex set in code selection the calculation is based by the 1992-1-1 DK NA:2007.



Shell, EC 1992-1-1 Annex F (Informative), Load comb.: 1

ID	Thickness	Elem	Node	Face	Sigma Edx	Sigma Edy	Tau Edxy	cot(theta)	F.1 (3)
[-]	[mm]	[-]	[-]	[-]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[-]	[-]
W.1.1	200	7	0	Bottom	0.02	0.37	-0.00	106.45	Reinforcement not required.
			0	Тор	-0.03	0.62	-0.00	390.16	Reinforcement required.
		8	0	Bottom	0.00	0.21	0.00	82.51	Reinforcement not required.
			0	Тор	0.00	0.79	0.01	61.07	Reinforcement not required.
		9	0	Bottom	-0.01	0.04	0.00	17.89	Reinforcement required.
			0	Тор	0.01	0.96	0.01	67.06	Reinforcement not required.
		10	0	Bottom	-0.00	0.21	0.00	105.78	Reinforcement required.
			0	Тор	0.00	0.79	0.01	150.20	Reinforcement not required.
		11	0	Bottom	0.03	0.44	0.01	40.23	Reinforcement not required.
			0	Тор	-0.05	0.56	0.01	88.38	Reinforcement required.
		12	0	Bottom	-0.12	-0.45	0.02	21.12	Reinforcement required.
			0	Тор	0.18	1.46	-0.05	27.39	Reinforcement not required.
		13	0	Bottom	-0.00	0.12	-0.01	21.01	Reinforcement required.
			0	Тор	0.00	0.88	0.02	35.64	Reinforcement not required.

Applied formulas	ftdx	ftdy	Sigma'cd	f tdx (Appl.)	f tdy	(Appl.)	nu	fcd	Sigma Rd
[•]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[N/	mm²]	[-]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]
-	-	-	-	0.96		0.96	0.562	10.7	5.99
(F5.); (F6.); (F7.)	0.03	0.00	0.62	0.96		0.96	0.562	10.7	5.99
-	-	-	-	0.96		0.96	0.562	10.7	5.99
-	-	-	-	0.96		0.96	0.562	10.7	5.99
(F5.); (F6.); (F7.)	0.01	0.00	0.04	0.96		0.96	0.562	10.7	5.99
-	-	-	-	0.96		0.96	0.562	10.7	5.99
(F5.); (F6.); (F7.)	0.00	0.00	0.21	0.96		0.96	0.562	10.7	5.99
-	-	-	-	0.96		0.96	0.562	10.7	5.99
-	-	-	-	0.96		0.96	0.562	10.7	5.99
(F5.); (F6.); (F7.)	0.05	0.00	0.56	0.96		0.96	0.562	10.7	5.99
(F2.); (F3.); (F4.)	0.14	0.47	0.03	0.96		0.96	0.562	10.7	5.99
-	-	-	-	0.96		0.96	0.562	10.7	5.99
(F5.); (F6.); (F7.)	0.00	0.00	0.12	0.96		0.96	0.562	10.7	5.99
-	-	-	-	0.96		0.96	0.562	10.7	5.99

F.1 (4) Appl.	ftdx (F.8)	ftdy (F.9)	Sigma cd (F.10)	F.1 (4)
[-]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[N/mm <sup>2</sup> ]	[-]
-	-	-	-	-
Not fulfilled!	0.03	0.03	0.66	Not fulfilled!
-	-	-	-	-
-	-	-	-	-
Not fulfilled!	0.01	0.01	0.05	Not fulfilled!
-	-	-	-	-
Not fulfilled!	0.00	0.00	0.21	Not fulfilled!
-	-	-	-	-
-	-	-	-	-
Not fulfilled!	0.05	0.05	0.61	Not fulfilled!
Not fulfilled!	0.45	0.45	0.34	Not fulfilled!
-	-	-	-	-
Not fulfilled!	0.00	0.00	0.13	Not fulfilled!
-	-	-	-	-

# 7. Steel design

## 7.1. Fire design

The aim of this feature is to check and design steel bars for fire effects according to EN 1993-1-2.

The feature can be found in *Steel design/Steel bar, fire design*. It needs some new input data of the bars, and a special load combination and/or load group must be defined.



In *Beam/Column /Material/Application data* there is a new safety factor  $\gamma_{M,fL}$  required for fire design calculations.

eam/Column		X
A.1 General Sec	tion Materia	End conditions
Library	Appl	ication data
Concrete 	Gamn Gamn EN 1	na M0 (U   Ua,Us) . 1.10 1.00 na M1 (U   Ua,Us) . 1.20 1.00 na Mfi (Ua) 1.00 993-1-8 specific: na M2 (U   Ua,Us) . 1.35 1.00
	Gamn	na M5 (U   Ua,Us) . 1.10 1.00

A "+Fire" type load case has to be defined in the Load cases dialog.

No	Name	Туре	Duration class (EN 1995 1-1)		ОК
1	dead	+Struc. dead load	Permanent		Cancel
2	live	Ordinary	Permanent		
3	fire	+Fire	<ul> <li>Permanent</li> </ul>		Import / Export
		Ordinary +Struc. dead load +Soil dead load +Shrinkage		-	Insert
		+Camber sim. +Fire			Delete
		+Seis load, FX+MX +Seis load, FX-MX +Seis load, FY+My +Seis load, FY-My			Delete all

For maximum of load groups calculations, an accidental load group must be defined that contains the "+Fire" type load case.

No	Load group	Included load cases	~	OK
1	permanent (Permanent, 1.00, 1.35, 1.00, 1.00, 0.85	dead		
2	temporary (Temporary, 1.50, 0.70, 0.50, 0.20, L)	live		Cancel
3	accidental (Accidental, 1.00)	fire		

For maximum of load combinations calculations, accidental load combinations must be defined that contains the "+Fire" type load case.

Load combination						
No	Name	Type	Factor	Included load cases	^	OK
1 test		Ua	1.00	dead		
			1.00	live		Cancel
			1.00	fire		
						Import / Expor

The effects (internal forces) are calculated from accidental load combinations, where fire is the accidental effect. Resistance of bar is calculated by using reduced yield strength and elasticity modulus for steel at the elevated temperature.

#### Steel design tab

A new design type can be found on the Steel design tab: "Steel bar, fire design".

It contains Calculation parameter, Check, Design group, Auto design and Manual design commands.



#### Calculation parameter

Explanation of data in calculation parameters are in EN 1993-1-2:3, 4 and EN 1992-1-2:3. "Deflection criterion is essential" option is available for Danish national annex only.

#### <u>Check</u>

It works in exactly the same way as in case of steel bar design.

#### <u>Design group</u>

It works in exactly the same way as in case of normal steel bar design, except that fire design parameter and fire design calculation parameter of two bars must match to be placed into the same design group.

#### <u>Auto design</u>

There are two design options:

- Design for fire protection material

The design parameter contains the fire protection material, which can be selected from a library (see later). Its *minimal and maximal thickness* and an increment value, which is used by the automatic design procedure to find the minimal necessary thickness of the protection material, can be given by the User

- *Calculate maximum temperature* The *Temperature step* for maximum temperature calculation can be defined by the User.

Limit utilization can also be set in the Steel bar - fire design parameter dialog
Oesign fire protection material	
Material	al wool, stone woo 🔻
Minimum thickness [mm]	1.0
Maximum thickness [mm]	100
Thickness increment [mm]	1.0
Calculate maximum temperature	
Temperature step [°C]	20
Limit utilization [%]	100

#### <u>Manual design</u>

Fire protection material (from a library, see later), its thickness, or the *Maximum member temperature* can be selected in Steel bar, fire protection dialog for Manual Design.

Steel bar, fire protection	×
Apply fire protection material	
Material Spray - mineral fibre	•
Thickness [mm]	. 10
Set temperature	
Maximum member temperature [°C]	400
ОК	Cancel

#### Fire protection material library

It is available by clicking on "Edit library..." item in the material list of fire protection parameter in Auto and Manual design dialogs.

Steel bar, fire protection	Fire protection material library
<ul> <li>Apply fre protection material         Material Spray - mineral fibre         Ubray - mineral fibre         Ubray - mineral fibre         Spray - vermiculite cement         Spray - vermiculite coment         Set temperat         Spray - vermiculite (or perite) and gypsum         Board - fore-silicate or fibre calcium-silicate         Board - fibre-silicate or fibre calcium-silicate         Board - fore-silicate or fibre-silicate, mineral wool, stone wool         Compressed fibre board - fibre-silicate, mineral wool, stone wool         Compressed fibre board - fibre-silicate, mineral wool, stone wool         Edit library         Edit library</li></ul>	Spray - mineral flore         Spray - vermiculite cement         Spray - peritie         High-density spray - vermiculite (or peritie) and cement         High-density spray - vermiculite (or peritie) and gypsum         Board - vermiculite (or peritie) and cement         Board - vermiculite (or peritie) and cement         Board - vermiculite (or peritie) and cement         Board - vermiculite (or perite) and cement         Board - sypsum         Compresed fibre board - fibre-silicate, mineral wool, stone wool         Concrete         Name

#### <u>Results</u>

Utilization results are available to display on the model and to list.

A new result "Max. of combinations/Bar, Combined utilization summary" is available, where the maximum utilization of steel bars for both normal and fire check is displayed

List tables		×
Eigenfrequencies     Seismic analysis     Foundation design     Load combinations     Max. of combinations     Maximum of load groups     Steel design     Input data     Load combinations     Max. of combinations	Max. of combinations          Data         Bar. Utilization         Bar. Fire design, Utilization         Bar. Combined utilization summary	List objects All  Send to Clipboard  Page setup Printer setup Vinter setup Settings > Add to batch List tables
Batch		Delete tables Save batch Load batch List batch
< Hide batch		Close

Max. of load combinations, Bar, Combined utilization summary

Member	Calculation	Maximum	Combination
[-]	[-]	[%]	[-]
B.1.1	ULS	121	Standard
	ALS (Fire)	98	Accidental

## 7.2. Steel joint stiffness

FEM-Design 17 gives the opportunity to set the rotation stiffness of *Steel Joints* automatically or manually, and apply it to the structure. This feature is available for *Steel Joints*, where at least one of the attached bars is connected with its end point by definition.

- 1. By clicking on the Steel design/Stiffness button, the *Steel joint stiffnesses* dialog appears.
- 2. Selecting a joint or joint group in the list will highlight the elements with thick red lines.
- 3. After that User can double click on the name of a joint/group in the list or just left click on the joint in the model to open the *Set joint stiffness* dialog.



Stiffness can be set only for previously calculated *Steel Joints*!

FD 17 uses the calculation method described in the Eurocode 3-8 for calculating joint stiffnesses. This method is valid only for joints with bolted endplate and column base of I section bars.

The program sets the stiffness of a joint by default as:

- hinged, if it can't take moment,
- semi-rigid with the calculated value, if it can take moment and stiffness can be calculated according to EC3-8,
- rigid, if it can take moment but the stiffness cannot be calculated according to EC3-8.

As the calculated stiffness value depends on the load of the joint, FEM-Design 17 by default uses the load combination that gives the highest utilization for moment. User however, can select another load combination, or can set stiffness manually.

) Hinged ) Rigid	⊙ Manu			m/°] vith highest ut	ilization				
Semi-rigid		-		elected by the					
Load combina	ation	M [kNm]	N [kN]	Util. (M) [%]	Sj [kNm/º]	Sj,ini [kNm/º]	Sj,ini/eta [kNm/º]	Use	1
SJ.11.1: 1.35*1 + 1.	.50*0.70*3 +	38	-54	47	1310	1310	437	Х	1
53.11.1: 1.35*1 + 1.	50~3 + 1.50	32	-45	40	1310	1310	437		

The details of the calculation can be found in the *Detailed result* and in the *Manual design calculation* dialog.

Moment resistance and stiffness - Beam 1 (EN 1993-1-8: [6.2.7]): 41 % (LC: 'SJ.1.1: 2')	Beam web transverse compression resistance
End-plate internal forces: N = -1.63 kN, T = 50.60 kN, M = -49.72 kNm	$b_{eff,c,wc}$ = 155 mm, $\omega$ = 0.90, $\rho$ = 0.77
T stub 1	$k_{wc} = 0.70$
Parameters and effective lengths at the end-plate	F <sub>o,wo,Rd</sub> = 188.32 kN
m = 40 mm, e = 31 mm, m <sub>2</sub> = 27 mm, p = 410 mm	Bolt-row capacities
l <sub>eff,cp</sub> = 254 mm, l <sub>eff,nc</sub> = 229 mm, l <sub>eff,1</sub> = 229 mm, l <sub>eff,2</sub> = 229 mm	F <sub>t1,Rd</sub> = 261.11 kN, h = 459 mm
g <sub>l.eff.op</sub> = 537 mm, g <sub>.eff.nc</sub> = 334 mm	$F_{t2,Rd} = 0.00 \text{ kN}, h = 49 \text{ mm}$
Parameters and effective lengths at the column flange	Joint moment capacity: M <sub>Rd</sub> = 119.85 kNm
m = 23 mm, e = 101 mm, m <sub>2</sub> = 20 mm, p = 410 mm	Stiffness calculation (EN 1993-1-8: [6.3])
l <sub>eff.op</sub> = 145 mm, ا <sub>eff.no</sub> = 236 mm, ا <sub>eff.1</sub> = 145 mm, ا <sub>eff.2</sub> = 236 mm	$\mathrm{S_{j}=64395}\ \mathrm{kNm/rad}$ , $\mathrm{S_{j,ini}=64395}\ \mathrm{kNm/rad}$ , m = 1.00
g <sub>l,eff,op</sub> = 483 mm, g <sub>l,eff,no</sub> = 331 mm	k <sub>1</sub> = 4.4 mm
Individual capacities	$A_{vo} = 3728 \text{ mm}^2$ , $\beta = 0.71$
F <sub>T,Rd,ep</sub> = 282.24 kN, Failure mode: 3	k <sub>3</sub> = 4.2 mm
F <sub>t,wb,Rd</sub> = 577.46 kN	k <sub>4</sub> = 29.0 mm
F <sub>T,Rd,of</sub> = 261.11 kN, Failure mode: 2	k₅ = 25.1 mm
$F_{t,wo,Rd} = 282.80 \text{ kN}$	k <sub>10</sub> = 7.0 mm
Final T-stub capacity: F <sub>T,Rd</sub> = 261.11 kN	L <sub>b</sub> = 56 mm
Column web shear resistance	z = 459 mm

Once the stiffnesses of the joints are set, when leaving design tabs the following dialog will pop up:



Selecting "Yes" will apply the new stiffnesses of joints to the attached bars, if "Apply stiffness to model..." box is checked in the *Set joint stiffness* dialog of the joint, or Apply column is checked in the *Steel joint stiffnesses* dialog next to the joint/group.

1310 398 1146	Yes Yes								
	Yes								
1146	100								
11-10	Yes								
"Rigid"	Yes								
		Set joint stiffness							
		Apply stiffness to model (when le	aving desig	in tabs or a	t Recalculate	e command)			
		) Hinged () Manu	al 1310	[kN	m/°]				
			of load on	mbination	ith higheet	tligation			
		Semi-rigid     Ouse S	j of load co	mbination s	elected by th	ne user			
		Load combination	M [kNm]	N [kN]	Util. (M)	Sj [kNm/º]	Sj,ini [kNm/º]	Sj,ini/eta [kNm/º]	Use /
		SJ.11.1: 1.35*1 + 1.50*3 + 1.50	34	-46	0	1310	1310	437	
		SJ.11.1: 1.35*1 + 1.50*0.70*3 +	41	-55	0	1310	1310	437	7 X
			Apply stiffness to model (when le     Hinged Manu     Hinged Manu     Seni-rigid Uses     Load combination	Apply stiffness to model (when leaving design     Hinged O Manual 1310     Hinged O Use 5) of load cor     Code combination     M     Load combination     M     S1.11.1: 1.15%1 + 1.15%3 + 1.50     34	Apply stiffness to model (when leaving design tabs or a     Hinged Manual 1310 [W     Miged	▲ Apply stiffness to model (when leaving design tabs or at Recalculate	Apply stiffness to model (when leaving design tabs or at Recalculate command)           Hinged         Manual         [1310]         [[btm/]]           Imaged         Imaged	Apply stiffness to model (when leaving design tabs or at Recalculate command)           Hinged         Manual         1310         [[bhm] <sup>o</sup> ]           Bigd         © Use SJ of fload combination selected by the user           Load combination         M         N         [[bhm] <sup>o</sup> ]         [[bhm] <sup>o</sup> ]           1.11.1:         1.35 <sup>2</sup> ± 1.150 <sup>15</sup> ± 1.150 <sup>15</sup> ± 1.150 <sup>15</sup> 34         ~6         0         1310         1310	Apply stiffness to model (when leaving design tabs or at Recalculate command)           Hinged         Manual         1310         [pkm/9]           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Imaged         Olse St of flaad carbination selected by the user           Image         Olse St of flaad carbination selected by the user           Image         Image         Image           Image         Image         Image         Image           Image         Image         Image

Before joint stiffness calculations	After joint stiffness applied on bar
Beam/Column X	Beam/Column X
A.1 General Section A Material H End conditions	A.1 General Section II Material H End conditions
Eccentricity in analytical model	Eccentricity in analytical model
The same at both ends Start End	The same at both ends Start End
Releases [kV/m, kVm/9]       Eccentricity [m]         e,x'       0.000         e,y'       0.000         phi,x'       0.000         phi,z'       0.000         phi,z'       0.000         mmm       mmm	Releases [dV/m, Mtm/9]       Eccentricity [m]         e,x       0.000         e,y       0.000         e,z       0.000         phi,x       0.000
OK Cancel	OK Cancel

## 7.3. Column base joint concrete tension failures

New verification is implemented for Column base joints that:

- calculates anchor forces and maximum concrete stress with nonlinear steel and concrete material behaviour,
- calculates capacities for adhesive and headed anchors for tension failure modes,
- works for both I and RHS sections.





The new parameters required for the new calculations can be found in Design tab.



Column Base plate		, b/2 , b/2 ,
Anchor bolt	3	b/2 b/2
Welds	-	
Foundation		a/2
- Material and		
Calculation		
- Anchor geor		a/2
Calculations		
		T
Data	Value	
Material	C25/30	
a [mm]	1000	h
b [mm]	1000	
h [mm]	600	+
ey	0	
	0	
ez	0	
ez u [mm] Decian	50	
	50 s d geometry parameters rete interaction metry	



Design Column Co	ameters e interac	tion		≥hmin	≥sn	nin ≥cmin +++	
Check requested	Yes	value					
Concrete is cracked	Yes						
gamma,Mp	1.50						
gamma,Mc	1.50	Ве	nded				
gamma,Msp	1.50	Data		Value			
kar	8.50	Check requested	Yes				
kucr	11.90	Concrete is cracked	Yes				
Ignore cone-failure	No	gamma,Mp	1.50				
Ignore splitting-failure	No	gamma,Mc	1.50		Head	led	
cmin [mm]	50	gamma,Msp	1.50	Data		Value	
smin [mm]	80	kar	8.50	Check requested		Yes	
hmin [mm]	260	kucr	11.90		_	Yes	
		Ignore cone-failure	No	gamma,Mp		1.50	
		Ignore splitting-failure	No	gamma,Mc		1.50	
		cmin [mm]	50	gamma,Msp		1.50	
< Back Joint lib	orary >	smin [mm]	80	kcr		8.50	
		hmin [mm]	260	kucr		11.90	
				Ignore cone-failure	. 1	No	
				Ignore splitting-fail	ure l	No	
				cmin [mm]		50	
				smin [mm]		80	

hmin [mm]

#### The picture below shows the output for adhesive (straight or bended) and headed anchorage.

#### Adhesive

Anchorage tension resistance (Ref.8: [2.4.1]): 86 % (LC: 'aa')

 $\begin{array}{l} \mbox{Anchor normal and shear forces} \\ F_{Ed,1}=-8.23 \ \mbox{kN}, \ \mbox{V}_{Ed,1}=15.81 \ \mbox{kN} \\ F_{Ed,2}=-56.82 \ \mbox{kN}, \ \mbox{V}_{Ed,2}=15.81 \ \mbox{kN} \\ F_{Ed,3}=62.29 \ \mbox{kN}, \ \ \mbox{V}_{Ed,3}=15.81 \ \mbox{kN} \\ F_{Ed,4}=13.70 \ \mbox{kN}, \ \ \ \mbox{V}_{Ed,4}=15.81 \ \mbox{kN} \end{array}$ 

Maximum concrete compression stress  $\sigma_{c,Ed}$ = -15.68 N/mm<sup>2</sup>

Adhesive failure resistance: 86 % Topology-independent anchor parameters I<sub>b</sub>= 300 mm, f<sub>bd</sub>= 2.70 N/mm<sup>2</sup> Dominant anchor: 3 Topology-dependent parameters for the dominant anchor c<sub>d</sub>= 105 mm,  $\alpha_1$ = 1.00,  $\alpha_2$ = 0.70 Normal force and capacity for the dominant anchor F<sub>Ed,3</sub>= 62.29 kN, F<sub>Rd,a</sub>= 72.71 kN

#### Headed

Anchorage tension resistance (CEN/TS 1992-4-2: [6.2]): 66 % (LC: 'aa')

Anchor normal and shear forces F<sub>Ed,1</sub>= -8.23 kN, V<sub>Ed,1</sub>= 15.81 kN F<sub>Ed,2</sub>= -56.82 kN, V<sub>Ed,2</sub>= 15.81 kN F<sub>Ed,3</sub>= 62.29 kN, V<sub>Ed,3</sub>= 15.81 kN F<sub>Ed,4</sub>= 13.70 kN, V<sub>Ed,4</sub>= 15.81 kN

Maximum concrete compression stress  $\sigma_{c,Ed}$ = -15.68 N/mm<sup>2</sup>

Pull-out failure resistance: 31 % Topology-independent anchor parameters An= 1649 mm<sup>2</sup>, f<sub>ck,oube</sub>= 30.00 N/mm<sup>2</sup>, ψ<sub>uor,N</sub>= 1.00 Dominant anchor: 3 Normal force and capacity for the dominant anchor F<sub>Ed.3</sub>= 62.29 kN, F<sub>Rd.o</sub>= 197.92 kN

Concrete cone failure resistance: 66 %

Dominant anchor group: 3, 4 Topology-dependent parameters for the dominant anchor group  $h_{ef} = 257 \text{ mm}, s_{or,N} = 770 \text{ mm}, A_{o,N}^0 = 592900 \text{ mm}^2, A_{o,N} = 680000 \text{ mm}^2$   $N_{Rk,o}^0 = 191.44 \text{ kN}, \psi_{s,N} = 0.93, \psi_{re,N} = 1.00, \psi_{ec,N} = 0.84$ Normal force and capacity for the dominant anchor group  $F_{Ed,3,4} = 75.99 \text{ kN}, F_{Rd,o} = 114.28 \text{ kN}$ 

 $\begin{array}{l} \label{eq:splitting failure resistance: 31 \% \\ Topology-independent anchor parameters \\ h_{min}= 260 \mbox{ mm}, \ s_{or,sp}= 780 \mbox{ mm}, \ A_{o,N}^{0}= 608400 \mbox{ mm}^2 \\ N_{Rk}^{0}= 241.91 \mbox{ kN}, \ \psi_{h,sp}= 1.75 \\ Dominant anchor group: 3, 4 \\ Topology-dependent parameters for the dominant anchor group \\ A_{c,N}= 685000 \mbox{ mm}^2, \ \psi_{s,N}= 0.93, \ \psi_{re,N}= 1.00, \ \psi_{ec,N}= 0.84 \\ Normal force and capacity for the dominant anchor group \\ F_{Ed,3,4}= 75.99 \mbox{ kN}, \ F_{Rd,sp}= 247.29 \mbox{ kN} \end{array}$ 

Blow-out failure resistance: 0 % Not relevant

The verification is carried out according to:

- The Swedish Institute of Steel Construction, Detail Handbook, Publication 183, 2.4.1 for the Straight and Bended type of anchors.
- CEN/TS 1992-4-2: [6.2] for the Headed type of anchors.

## 7.4. "Rotated" option for hollow sections

Hollow sections can be rotated separately in all joints where they are available.

Data	Value
Cross-section 1.	KKR 200x120x8
Material 1.	S 355
Rotated 1.	No
Cross-section 2.	KKR 200x120x8
Material 2.	S 355
Rotated 2.	No
a [mm]	0
b [mm]	1000

Data	Value
Cross-section 1.	KKR 200x120x8
Material 1.	S 355
Rotated 1.	Yes
Cross-section 2.	KKR 200x120x8
Material 2.	S 355
Rotated 2.	No
a [mm]	0
b [mm]	1000

Setup components automatically when cross-section changed

Data Value KKR 200x120x8 Cross-section 1. Material 1. S 355 Rotated 1. Yes KKR 200x120x8 Cross-section 2. Material 2. S 355 Rotated 2. Yes a [mm] 0 b [mm] 1000

Setup components automatically when cross-section changed

Eurocode (NA: Hungarian)

Eurocode (NA: Hungarian)

Eurocode (NA: Hungarian)







## 7.5. Joint library

Both in *Steel joint* module and in *Steel joint manual design* window of *3D Structure* module the customized joint types can be saved into/loaded from a library.



By clicking *Save joint*, the *Save joint to library*... dialog opens. In the dialog, in the bottom-left corner an arbitrary name can be typed for the joint. After clicking Save in the top-right corner, the saved joint appears in the library.



Using the buttons in the bottom-right corner, User can rename, copy, delete, import or export joints (from/to \*.fdljoint format).



To rename or copy an existing joint, firts type the new name into the textbox under the library, then click on *Rename*.



User can load previously saved joints by clicking on *Joint library/Load joint*. The *Load joint from library*... dialog opens, which is similar to *Save joint to library*... dialog, except that the *Rename, Copy* and *Delete* buttons are disabled, and instead of *Save* button, *Load* button appears.

	Load CB1 joint
	Save CB1 joint
	Edit CB1 library
ibrary >	

Joint li

To load an existing joint, select it from the library then click Load.



By clicking *Joint library/Edit joint* the *Edit library dialog* opens It looks the same as *Save joint to library*. Here the user can rename, copy, delete joints and import/export from/to the library.

	Load CB1 joint
	Save CB1 joint
	Edit CB1 library
Joint library >	

In *3D Structure* module at *Steel design tab/Steel joint/Define joint* User can select a joint from the joint library by checking the *Select joint from library* checkbox.



In the *Load joint from library* dialog, only those joints can be chosen thatare compatible with the selected bar(s). Those joints are shown active (black) in the library.

Load joint from library		
HE-A 300	]	-
HE-B 300		



If User selects a non-appropriate joint for the selected element(s), FEM-Design sends a warning and automatically creates a joint fitting for the selected item.





Each joint solution has to be saved to its own library. There is no possibility to save all joint types and solutions into one library.

## 7.6. Steel joints added to Filter

In 3D Structure module steel joints can be filtered either as Joint component, or by their identifier.



## 7.7. User interface improvements in Steel joints module

#### 7.7.1. Tooltip for joint bars

Tooltip for joint bars has been implemented for easier identification in *Steel joint* module and *Manual design* window in *3D Structure* module.



#### 7.7.2. Joint bar display option

The positive directions of loads can be displayed on the joint by checking the *Display positive direction of loads* option in the *Settings/Bars.../Display/Bars* dialog in the *Steel joint* module and in the *Manual design* window in the *3D Structure* module. The size of the arrows can be set, too.



#### 7.7.3. Navigation buttons for steel joints

In FEM-Design 17, while using Steel joint Manual Design in *3D Structure* module, User can switch between *Previous* or *Next* joint by clicking up and down arrows next to the drop-down list.

This feature can be found in *Steel Joint* module, too.

Joint			
1.1		+	♠ ↓ .
	Collection		

## 7.8. Steel joint utilization in Documentation

Steel joint utilization can be listed under Steel design/Load combinations/Joint, utilization.



## 8. Results

#### 8.1. Dominant load combination shown on results for maximum of load combinations

*Maximum of load combination results* now shows the ordinal number of the load combination, from which the min/max value comes from. If User holds the cursor over a point, the ordinal number and the full name of the load combination are shown in a tooltip.



The display of number of load combination can be hidden by unchecking the 'Shoe extra information, if available' box in the Numeric Value dialog.

#### 8.2. Displacement-like results improved

*Displacements* result's name has been modified to *Translational displacements* and a new result called *Rotational displacements* is added.



Also colour-palette option is added to *Translational displacements* results that allows for better presentation and understanding of displacements results in some cases.

Results ⊡- Analysis	Translational displacements
<ul> <li>Load cases</li> <li>Translational displacements</li> <li>Rotational displacements</li> </ul>	Load case type U - Ultimate     Display options
···· Reactions ···· Connection forces	Load cases Graph © Colour palette
Maximum of load combinations	

User can find a few powerful options to visualise the displacement results in *Display options* dialog. They give many possibilities to show the result in any pre- or user defined directions. The pictures below present visualisation possibilities.













The Auto apply can be turned off in case of huge models, to avoid long-time refreshing.

#### 8.3. Shear center result

In FD 17, the shear center can be displayed for every storey.





Each displayed shear center represents the result of a calculation based on the storeys below that storey. For example, the calculation of the center displayed on "Storey 2" takes also "Storey 1" and "Foundation" into account.



Shear center results can be listed in *List tables dialog/Analysis/Eigenfrequencies/Shear center*.

List tables			×
Tables Structure Loads	Ê	igenfrequencies	List objects
Coads     Finite elements     Quantity estimation     Analysis     Coad cases     Moving load maximum     Imperfections     Load combinations     Max. of combinations     Maximum of load groups     Stability analysis     Stability analysis     Seismic analysis     Foundation design		Result Nodal masses Total/Level masses Shear centers Eigenfrequencies Nodal vibration shapes  Shapes  1. 2.	<ul> <li>AaBbYyZz →</li> <li>Hide irrelevant tables</li> <li>Regional decimal symbol instead of dot</li> <li>Fill all table cells</li> </ul>

Shear centers

Alpha	EI1	EI2	Х	У	Z
[rad]	[kNm2]	[kNm2]	[m]	[m]	[m]
-1.571	885272790.059	540718616.178	16.000	20.000	3.000
-1.571	3449176940.867	1528795003.672	16.000	20.000	6.000
-1.571	6326495388.797	2475347677.958	16.000	20.000	9.000
-1.571	9170982655.342	3379707268.708	16.000	20.000	12.000

### 8.4. Mass result

In FEM-Design 17, Masses result is renamed to Nodal masses, and new result types: Nodal masses and Total / Level masses are available,

Loads:



#### In FD 16:

Display result		×
Results - Analysis - Eigenfrequencies - Masses - Vibration shapes	Masses	

Eurocode (NA: Hungarian) code: Eigenfrequencies - Converted masses - [t]







Eurocode (NA: Hungarian) code: Eigenfrequencies - Converted masses - nodal - [t] View: 2

Eurocode (NA: Hungarian) code: Eigenfrequencies - Converted masses - total/level - [t]



## 8.5. Minimum and maximum values of soil results in colour palette view

In the earlier versions, the results (translations or stresses) of solid (volume) elements in colour palette showed only the results on the sidewalls of the volume.

In some cases the **minimum/maximum values are inside the volume** (not on the surface) therefore, now the minimum/maximum **points are indicated**, and are showing the positions (coordinates) and the minimum/maximum values. Using these points: User can for example make a 2D view section where the minimum/maximum value can be visualised.

This new feature is available for *Translational displacements* or *Solid stresses* results in *Display options/ Default settings/ Others* dialog. User can set the font size and type, too.

📰 🕂 📬 Display options 📿 🗖 📬	11:10:10 🕻 🖧 🏶 🕁 🚳
Cisplay options         I t t t         I i i i i i i i i i i i i i i i i i i i	Solid, colour palette         Palette         Display with deformed shape         Scale factor         Scale factor         10,101848 mm         Reset absolute         Reset relative         Colour         Pen width for bars [mm]         Pen width for bars [mm]         Solid         Show solid minimum and maximum         AaBby         Font formatting

Result displaying auto min/max stresses	Result displaying auto min/max translations	2D view through the maximum point
Constitution 1:3) Transmission de Competition (ref 0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,	Events note: Los dens-1-(2): Transitional disconsense-Aux- Color-pather - (net)	Excede code: Load cases - 1 - (1) - Trankitorial digitoconetts - Alts - Colour palette - [rm]

#### 8.6. Local stability results with more details

The maximum value and detailed calculation of the result is displayed.

Eurocode (NA: Hungarian) code: 1st order theory - Load combinations - U - Local stability - Overturning of walls - [%]



Eurocode (NA: Hungarian) code: 1st order theory - Load combinations - U - Local stability - Sliding - [%]



The detailed calculation can be hidden by unchecking the 'Show extra information, if available' box in the Numeric Value dialog.

#### 8.7. Colours of principal stresses, moments and normal forces

Colouring of principal stress, moment and normal force results of shells has been modified to make the result more clear. The algebraically **larger value** will be **the first principal value**, by default indicated with green colour. The algebraically **smaller value** will be **the second principal value**, by default indicated with red colour. Difference in the *Display options* dialogs and the results is shown on the picture below.



Eurocode code: Load cases - d - (U) - Shells, Sigma 1/Sigma 2, top - Direction - [N/mm2]



*Display options* dialog for principal stresses of solids has been modified in a similar way as for shells, but in this case, three colours can be set for the three principal stresses.

Arrows at the ends of the lines are indicating whether the principal stresses are tension (positive) or compression (negative).

One size [m] 0.500		$\leq$	
Proportional - scale     0.0017	74		
Scaling value 2.871e-01 N/	mm2		
Reset absolute Reset relativ	/e		
	_	3	
Colour 1			11 - 0.0000 March 2
Colour 2			(*************************************
Colour 3			
Pen width [mm] 2	×		
Settings > OK Can	ncel		

### 8.8. Option to hide finite elements for colour palette results

To hide the finite elements from the result, click on the icon. In the pop-up window, *Surface, colour palette* click the *Others* tab and uncheck the box next to *Show finite elements*, then click 'OK'.

Picture No. 3. shows the result without the finite elements grids.



Eurocode (NA: Hungarian) code: Load cases - 1 - (U) - Translational displacements - Abs - Colour palette - [mm]

## 8.9. Scale to view option for colour palette results

In FEM-Design 17 for results displayed by colour palette, a new option is available in the *Display options* dialog. It automatically rescales the results considering the maximum and minimum values of the current view.



With *Scale to view* option turned ON, if User hides part of the structure, the palette will be rescaled automatically:





If the result to be displayed needs wider scaling than the customized has, the program overwrites the scaling to the view.

## 9. Documentation

## 9.1. Option to hide sections

In order to speed up work in Documentation, Sections can be hidden/shown by clicking the checkbox next to the section's name.

The items' titles of the hidden Sections are listed with grey color and their contents are not visible in the documentation. Only a comment on a single page shows that the Section is hidden. Consequently, if a Section gets hidden, the page numbering of the Sections following it, changes.



The hidden sections are not locked thus, they can be cut, copied, pasted, moved, dropped or modified. If an active section is moved into a hidden one, both sections will be hidden and vice versa.



It is possible to select more sections to hide at one go. Uncheck an active checkbox, all the selected sections will turn off. Check an inactive one, they will all be shown.

## 9.2. Append templates

From now on, User can append previously saved Templates in Documentation. Items of the appended templates will be inserted after the last existing document item.

The Append function can be found at the Templates menu in the bottom-left corner:

The Descript 17: 28 Section 4: U.S.Co     Terror Markov Markov Markov Markov Markov Markov     Terror Markov     Terror Markov     Terror Markov     Terror     Terror     Terror     Terror     Terror     Terror			
	Navig.		* 210 <b>7</b> 2 100
	Ite	ms	
	Cut	Сору	
	Paste	Delete	
00	Properties		
		Apply	
Normation	Docu	Append	
Brens Dat Cape	Reset	Save as	
Prov Deles Provision Decement Rest Pro.	Templates >	Save as .docx	1
Templates > Save as .docx		· · · · · · · · · · · · · · · · · · ·	

## 9.3. Docgraph windows are created according to the saved Window settings

User can save the Window settings for Docgraphs as default by clicking the *Save as default* button in the *Settings* dialog.



Pictures 1 and 2 show that, if User turns off the grid but doesn't save that setting as default, the next inserted docgraph will be displayed as the first one was displayed originally.



If User saves the customized settings as default, every next inserted docgraph, will be displayed according to the saved properties. (See picture 3. and 4.)



## 10. Others

## 10.1. FEM-Design Center

FEM-Design systray has been removed and replaced with a new, more informative tool called FEM-Design Center. All the modules, useful links, updates and links to the add-ins are now in one place.



At the end of FEM-Design installation, user can select an option to launch the FEM-Design Center and pin it to the taskbar for easier access, if necessary.





CTRL+Left mouse button will open the last model.

Shitf + Left mouse button will open history in a jumplist.

We highly suggest the User to pin the command center so right clicking the icon on the tray will open a jumplist where all the modules are listed.



## 10.2. Drag and drop

In FEM-Design 17 "drag and drop" function is implemented. Files in all supported formats can be now simply dragged and dropped to the FEM-Design.



#### **10.3. FEM-Design Support tool**

This tool helps the User to reach the Support team easily and get answers for specific model dependant questions. This can be found at *Help/FEM-Design Support* menu.

FEM-Design 17 - 3D Structure - Untitled								
File Edit Draw Modify Tools Settings Vi	ew Window	Help						
Structure Loads Finite elements Analysis	Foundation de	FEM-De	sign Support					
Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid         Image: A structure grid       Image: A structure grid       Image: A structure grid	Bar	FEM-De FEM-De FEM-De Downloa Check fr	-					
🛞 FEM-Design Support			- 🗆 X					
StruSoft       I prefer answer via email       Help         Image: Construction of the system o								
		<b>^</b>	Add files Remove file					
S								
Include your contact information. Strusoft Support will con Name	Telephone	as possible.						
Email address*	Company name	•						
Description								
			^					
			~					
			Send to Support					

User can decide how to receive answer, via email or phone and/or TeamViewer (remote desktop

If the *Attach current model* is active, the current, saved model will be send to the support along with the report. If the model includes linked DWG files, those files will be attached as well.

Add Files allows to add more files (e.g. pictures) to the report.

The *Help* button links to a guide on how to use this tool. There, one can find more detailed explanation of the tool functionality and its limitation.

## 10.4. Company settings

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It is a very fast way to export all FEM-Design settings and libraries to any other computer. With this feature User can deliver their company preferences from one computer to another very easy and quickly.

StruSoft FEM-Design 17	– 🗆 X
FEM-Design 쭇	StruSoft
3D Modules         Image: Structure str	<ul> <li>What's new?</li> <li>Support</li> <li>Manuals</li> <li>Webinars</li> <li>Trainings</li> </ul>
2D Modules	<ul> <li>Forum</li> <li>Videos</li> <li>Release notes</li> <li>Check for update</li> </ul>
	<ul> <li>G Check for upgrade</li> <li>↓ Download center</li> <li>○ Revit Add-In</li> <li>○ Tekla Add-In</li> </ul>
Export Import	

Export and import settings can be found in the FEM-Design Center. The *company settings* file has *\*.fdcs* format.

FEM-Design Company Settings	-	_		$\times$
Settings and libraries Settings Material library Section library Reinforcement library Bolt library Predefined load library Timber panel library Joint libraries Fire protection material library	Additional files - release \templates \fdacelkeret.bsc - release \templates \fdacelkeret.xltx		Export Close	
	<	>	Files <u>A</u> dd <u>R</u> emove	e

The following items can be exported:

- settings
- libraries
- doc templates
- list batches
- title blocks
- list templates
- Office OpenXML templates

## 10.5. Graphic engine settings and Draft mode

The graphical settings moved to Environment/Graphic engine and two font rendering option is available. By using the Draft mode, drawing the texts to the screen is much faster but the quality is bit lower.



# Version 16

# Version 17

## 10.6. Timesave

Timesave (backup) in FEM-Design 17 will save only the input file and not the result file anymore. This ensures a quite fast backup of model using less disk space. There is no option to save the result file in case of timesave. To save the result file, please *Save* the model from *File/Save* or (Ctrl+S).